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THE SPACE BASED MOON ORE CAPTURE SYSTEM
AND
MANUFACTURING PLANT

Final Report

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Since the early 1960's, when von Braun conceptualized his donut-shaped space station, men have dreamed of building a similar space station. But because of our limited present and future lift capabilities and because of the enormous energy and cost requirements necessary to dice the needed material out of the earth's atmosphere, a large orbiting space station has remained a virtual impossibility.

This design project, the Space Based Moon Ore Refinery and Manufacturing Plant, proposes to: 1. Design a space structure capable of receiving raw materials; 2. Process the raw materials into useful metal products; 3. Utilize these materials in order to build a large space station in a high earth orbit at a substantially lower cost than has been available.

It will be necessary to use a magnetic rail launcher to launch the raw materials from the moon's surface. It's trajectory must pass through the lower earth's atmosphere which will provide the needed atmospheric drag to slow the ore modules sufficiently so that the modules can be intercepted by the Capture System. A detailed structure analysis of this collision was made by using GIFT3, a commercially available finite element structural analysis program.

After the ore has been captured, it will be transferred to a manufacturing plant in a near earth orbit. Since the moon ore catcher is the primary design concern of our project, a previously designed manufacturing facility was used in conjunction with generic habitats. Design criteria for the plant was taken from Gerald Diggers(Space Applications Inc.) and Jonathan Newman(Amherst College) who conceptualized a detailed protoplant.¹ This refinery and manufacturing plant will provide the necessary means to produce refined materials. These products derived from raw materials found on the moon will, in

turn be utilized in the construction of a large space station in a geosynchronous Earth orbit.

The capture system will require several support systems to include power, thermal protection, attitude reference and control, communications, propulsion, and orbital maintenance. A power system will consist primarily of a solar array and storage batteries. The storage batteries will be very important as a power source during the maneuvering of the capture system while receiving the moon ore. Thermal protection will be limited to a passive system required to protect the electronic equipment. The attitude reference and control system must be very accurate in order to properly intercept the moon ore in its trajectory and deliver it to the manufacturing plant. The communication system will not be very extensive in that it will relay all of its necessary information and system status through the manufacturing structure. A propulsion system will be needed to enable the capture system limited maneuverability in order to receive the ore as well as to maintain the required orbit.

In addition, the manufacturing plant itself will require similar support systems. The large power requirements of the plant will be met using a large solar mirror. Additional power may be obtained through the utilization of some of the by-products of the manufacturing process itself. The structure of the plant will consist of a modular habitat and working area, built around the actual furnace, that will contain the crew and the other support systems. Thermal protection will require both active and passive systems in order to maintain proper conditions for both the crew and the electronic systems onboard. The attitude reference and control system will be a very elaborate combination of a number of systems that

will be able to precisely determine the correct attitude. These systems will include horizon sensors, star cameras, gyros, and sun sensors as well as control systems needed to adjust the attitude of the plant. A sophisticated communications system will be needed to ensure proper relays between the electromagnetic rail station on the moon and the plant. The trajectory of the moon ore must be constantly monitored and transmitted to the capture system as well as being able to communicate with earth stations the status of the plant and accompanying systems. The manufacturing plant will not require any propulsion system except for that necessary for attitude and orbital corrections. Since the structure will be in a low earth orbit and its mass will not be constant, there will be a need for cold gas jets or some other system to maintain the proper altitude.

Once the complete system is constructed and operational, the only required earth-based service will be to help monitor operations, help to shuttle new crews to the plant, bring food and other living necessities, and occasional parts for repair. For the most part the capture system/manufacturing plant should be able to operate independent of earth-based facilities as it will have the capabilities to make most of its own repairs and to manufacture some necessary consumables like hydrogen and oxygen which are by-products of the manufacturing process.

In designing the LEO capture system and the accompanying manufacturing facility a number of prerequisites were established. Among these, obviously, was the ability to catch the ore modules in rapid succession over short time intervals with a minimum maintenance requirement. In order to successfully capture the moon ore, the relative kinetic energy between the incoming ore and the capture system had to be

reduced to a minimum before impact, thus minimizing the energy dissipation requirements of the capture system itself. Due to orbital perturbations and possible deviations in the ore's trajectory, the capture system must have a large receiving area and have some degree of maneuverability. After successful capture of the ore, it must be transferred (in a timely manner) to the manufacturing facility where it will be processed into a usable form. As should be expected, there are significant power requirements on the manufacturing plant that must be met. To allow for the continuous monitoring of the manufacturing process, accommodations for the crew must be made available, preferably in the form of an adjoining habitat.

In implementing the entire system, assumptions regarding the projected level of technology must be made. One of the assumptions include the ability to establish and maintain a moon-based electromagnetic launching facility which will have the capability to cut the moon ore in an acceptable trajectory. Furthermore, the ability to compute and track this trajectory is necessary. The ability to place the capture system and manufacturing facility in its desired orbit will be dependent on the capabilities of heavy-lift launch vehicles and Shuttle Derived Vehicles now in development.

Based on the aforementioned prerequisites a few possible solutions came to light in the initial design phase. Through the process of elimination, the team soon focused on a single concept and the design became an optimization problem.

One of the possible solutions the design team considered was a large conical shaped cylinder with internal baffles to slow and stop the incoming ore. This concept was taken from *Space Based Manufacturing*

Team Masterminded's Design: It would be able to take in the material as it is large, and it is assumed that some type of processing system could be added to provide it with some maneuverability. The drawbacks of this design, which took it out of consideration were that it was not versatile enough and the design of the baffles was too difficult. Its lack of versatility was marked by its inability to handle the addition of other subsystems. The construction and repair of a facility like this would also be much more complicated than in the case of other proposed ideas.

Another idea that was considered involved sending huge containers full of moon ore. The containers themselves would be equipped with an active on-board system that could provide a large increase in drag or even with an active maneuvering capability. There were several inherent drawbacks to this design, including: a phenomenal power requirement to launch it from the moon, a large increase in the amount of equipment needed on the moon, and exorbitant costs to finance and maintain the active systems aboard the ore packages.

Another of the team's proposals was the forerunner of the actual system developed. It consisted of a large net stretched over a very large area with a limited depth support structure. The structure was very simple in design and had very little dampening incorporated into it. In addition, its attitude control functions would have to have been very extensive due to the instability of the system. This system did, however, provide an acceptable baseline on which to develop the existing design. This basic design is shown below in Fig. 1.

The structure of this initial design was enhanced in a manner which solved many problems. The size of the netting support structure was increased, causing three significant improvements. First, it allowed for

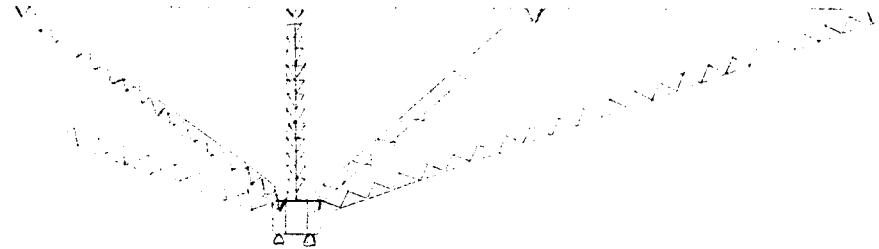


Fig. 1

an efficient dampening system to be placed on board. Secondly, the greater depth minimized the attitude control and stability problem. Lastly, the larger structure provides an excellent foundation for many possible subsystems.

As stated earlier, the ore modules will be launched from the moon's surface by means of a magnetic rail launcher. The ore will escape the moon and enter into a Hohmann transfer orbit which will send it on its way to Earth. During this time the ore is travelling at great speed and thus has a great deal of energy. It is necessary to reduce the velocity of the modules for the capture system interception. This could be accomplished any number of ways including many active systems such as drag producing balloons or some type of propulsive system. Although effective, these options are too expensive and would destroy the economic advantage this entire system is designed to achieve. The design group decided to use a passive dragging technique in that we would send the ore modules through the earth's atmosphere on its path to the capture system. There is no cost involved and the heat generated by the passage will be dissipated on the outward trip to interception at apogee. Figure 2 illustrates the path of the ore modules from start to finish.

The capture system primarily consists of four different components. The most significant and most noticeable of these components are the eight large arms extending radially from the center out. These will be

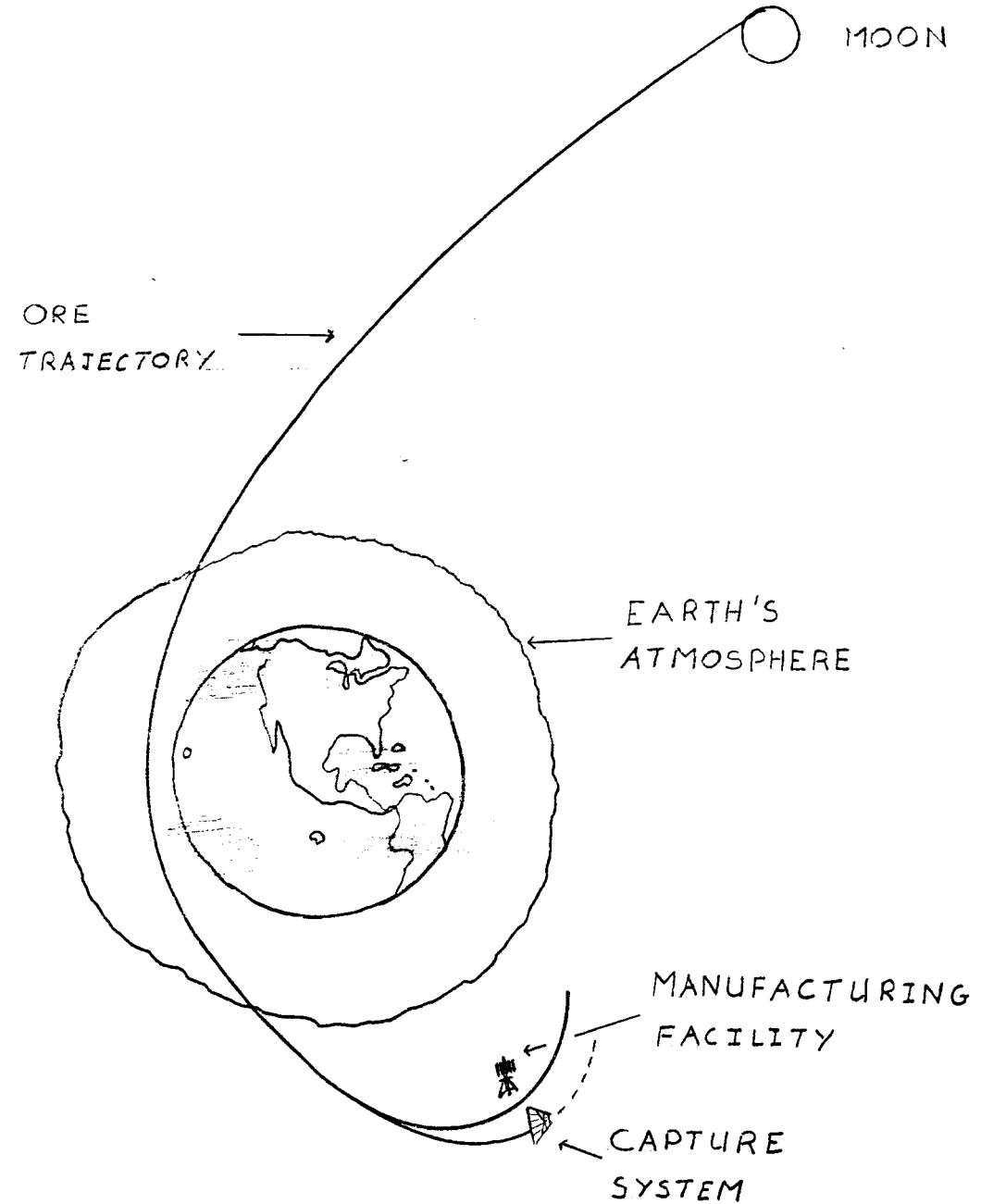


Fig. 2

referred to as A-beams. Providing structural support to the A-beams also extending from the center out are the smaller C-beams. The C-beams are connected at one end to a support on the center-structure and the other to a point on the A-beam. The center-structure will not only act as a common connection point, but will house the command and control center for the capture system. The fourth and most important component of the capture system is the netting structure. Connected to each of the eight A-beams at their tips and spanning the entire frontal cross section, the net will provide the stopping barrier for the incoming moon ore, thus allowing removal and transfer to the manufacturing plant and preventing any damage to the A-beams, C-beams, or the center-structure.

A front and side view of the command and control center are found below in Fig. 3.

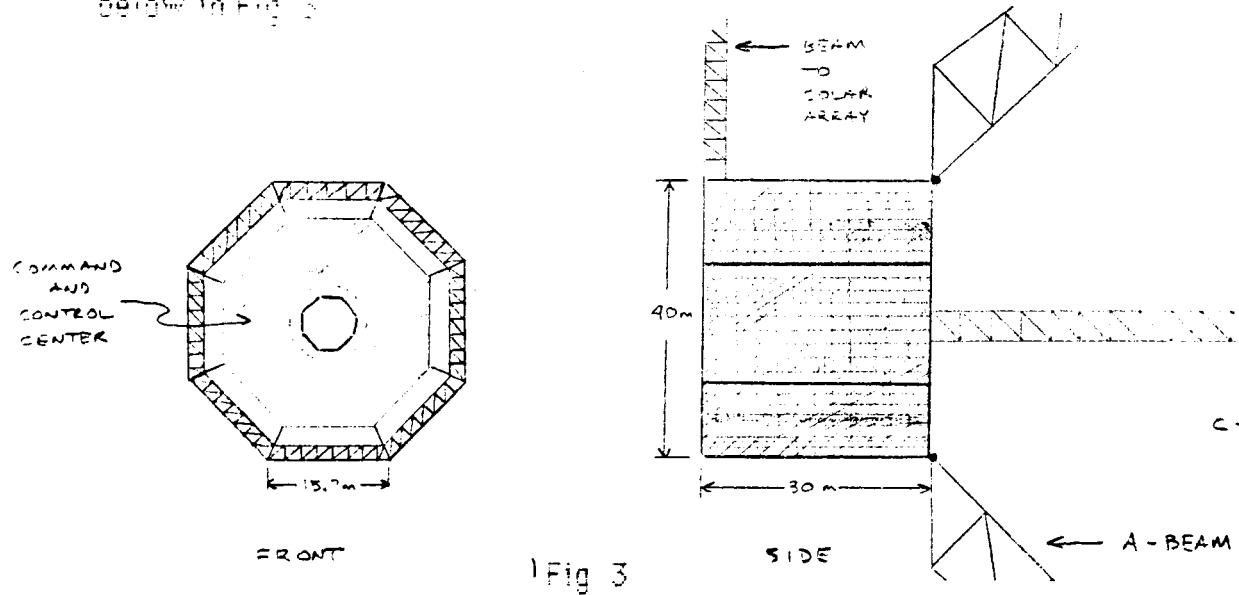


Fig. 3

This structure is an octagonal cylinder 40 meters in diameter and 30 meters deep. Each straight side of the octagon is approximately 15.7 meters in length, thus allowing adequate room to mount the base of each A-beam. On board the command and control structure will be a suitable zero-gravity working habitat containing working spaces for a crew of five.

to six - a controlling computer with two deck sub-computers, an environmental control system, and a propulsion system mounted at the base. Also mounted on the base is an extending boom with a solar array to provide power to on-board batteries. Note from the sketch of the C&C center the arm extending toward (what will be) the center of the capture system. The free end of this arm will provide the connection points for the C-beams.

As previously mentioned, the base of the A-beams will be connected along the straight side of the C&C octagon. The base cross-section of the A-beam is configured as an isosceles triangle with a 15 meter side length (See Appendix A). One side of the isosceles triangle will be hinged to the C&C structure. This movable hinge allows for ease of deployment of the arm itself and the netting. In Appendix A it is seen that the A-beam tapers from a 15 meter side length base to a 2 meter tip. This modification leaves a great deal of weight and, in turn, money. Each element of this beam is a rod measuring 0.5 square inches in area. It is made of an aluminum alloy of the characteristics given in Appendix B.

The C-beam is a simple triangular shaped beam with no taper. As shown in Fig 3, it is connected to the center arm using a hinge connection. This hinge connection allows a distinct advantage. If the C-beam is connected to the A-beam with a moveable connection, the angular position of the C-beam will determine how wide the frontal area of the capture system becomes.

Figure 4 depicts a simple design for a movable connection between a C-beam and an A-beam. The design of the C-beam complete with dimensions is found in Appendix C. The netting that is to be stretched across the front of the capture system will be made of Kevlar 49 material.

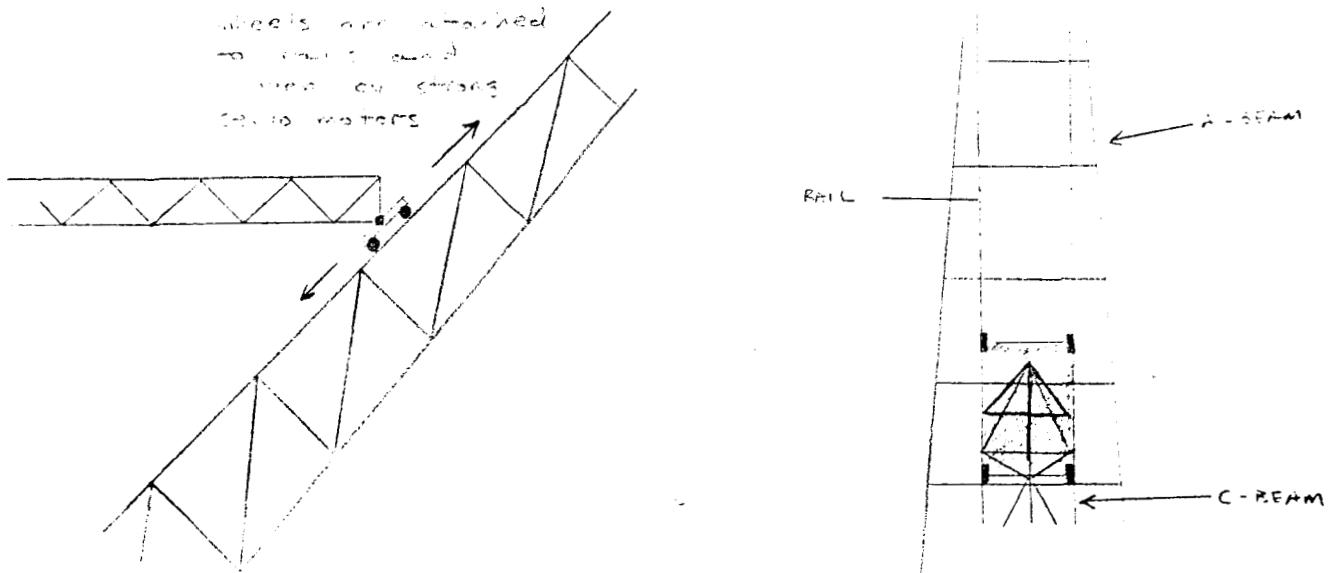


Fig. 4

It has the following properties:

$$\text{Density} = 1.45 \text{ Mg/m}^3 \quad \text{Strength} = 3.6 \text{ GPa} \quad \text{Modulus} = 130 \text{ GPa}$$

Based on these parameters, the cross-sectional area of the net webbing only has to be 0.2 square inches at any impact point. The actual allowable area of impact is a circle of radius 142 meters centered on the frontal projection. The webbing outside this area must be somewhat stronger to dissipate more of the energy quicker should some moon ore hit off target. The cross hatch pattern of the net depends on the size of the modules to be caught. The larger the modules, the less Kevlar needed but stronger beams would be needed. Thus, there is a trade off.

The Kevlar net will only be attached at the tip of each A-beam by one cable. This cable will be able to sustain the maximum amount of tension expected without failure. The cable will not be rigidly attached to the A-beam but rather it will be bent over a pulley of some type at the tip and then connected to an active damping system located somewhere along

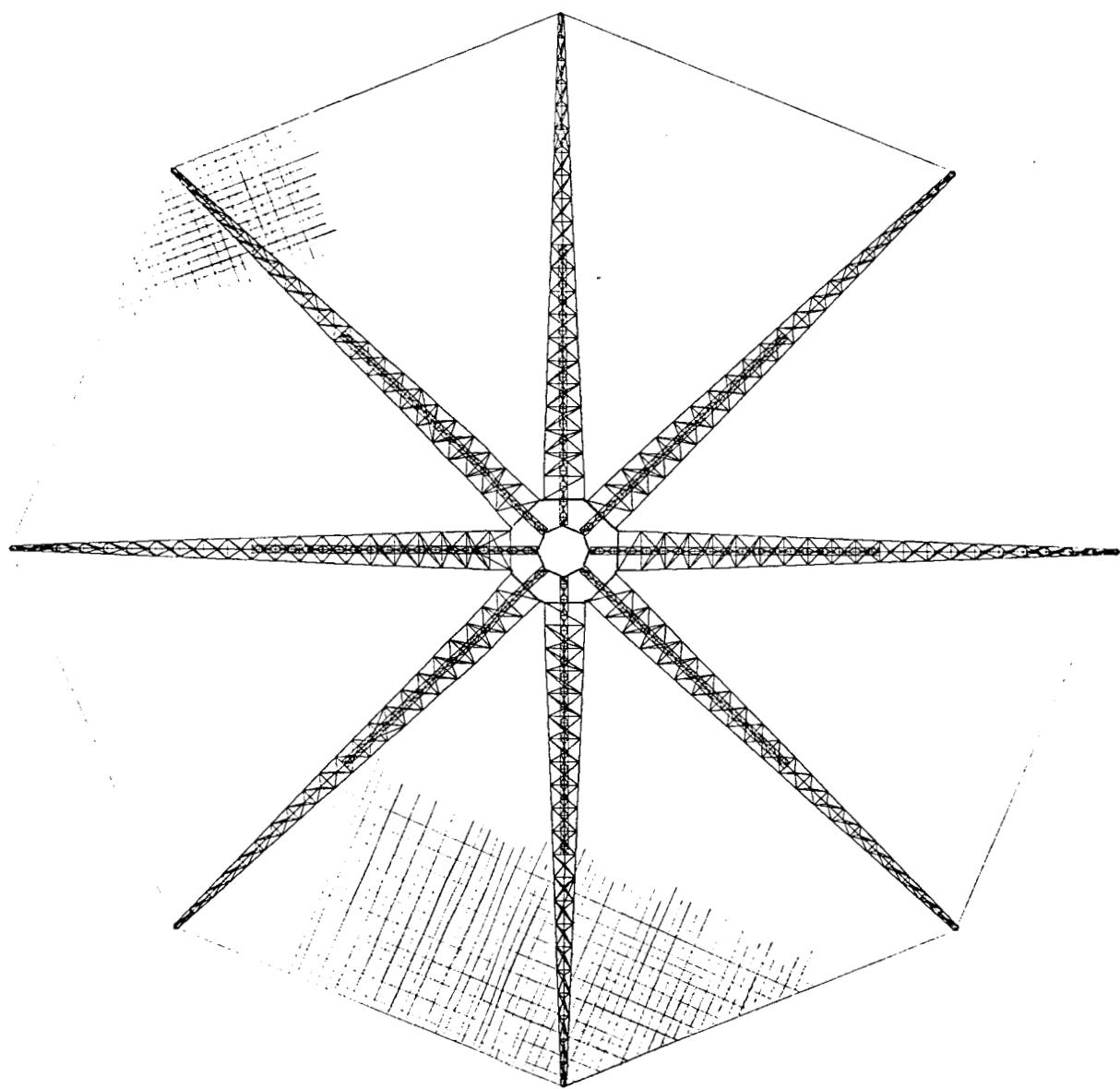
the A-beam. The deflections will have the same effect as the three wires do on an aircraft carrier.

The lengths of the beams (A-beam and C-beam) were basically determined through trial and error. The only definite parameter was the area of capture. The diameter of the frontal area had to be 400 meters. Considering the maximum allowable deflection of the net, the maximum angular deflections of the C-beams and A-beams, and keeping the amount of material to a minimum gave the present dimensions found in Figures 5 and 6.

The structural analysis of the capture system was done using the GIFT5 computer program. Only one of the structure's eight arms was modeled because of the symmetry which existed. Two load cases were modeled to simulate the largest load conditions which the capture system structure would encounter.

The modeling process began with the building of data files for each of the trusses used. The programs "ABUILD" and "CBUILD" found in Appendix D generated the data files for the inner and outer trusses, respectively. The data points for these trusses are located in files "BEAMA" and "BEAMC", also located in Appendix D.

The program "IBBUILD" was then created to build a "BULKM" driver file. The driver file is called "PR0M.SRC" and contains all responses and data necessary to execute the "BULKM" module of the "GIFT5" program. The program "IBBUILD" also creates a polygon file which draws all eight arms of the structure using the VAX computer and PS300 terminal. The data file containing this information is called "PROJ.PF". Neither "PROJ.PF" or "PR0M.SRC" is listed in the Appendix because of the large number of data points and connection parameters contained in each, which



100 meters

Fig. 5

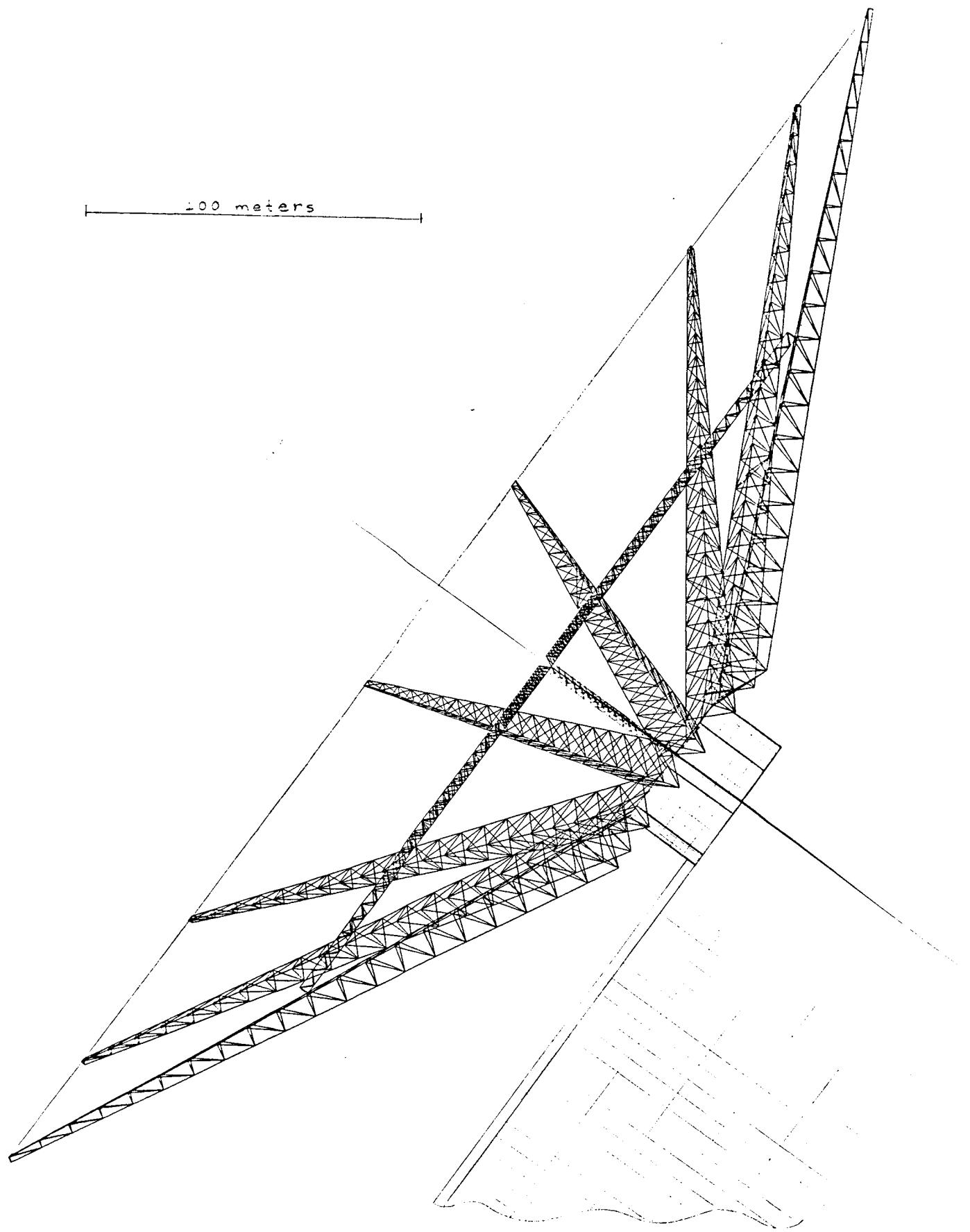


Fig. 6

the relatively meaningless to the reader. In a similar fashion the program "GIFTSLD" creates a "BULKLB" driver file for the GIFTB program. The driver file is "PUSH.SRC".

The program "GIFTSRUN" is performed to execute the entire static analysis of the capture system. The user is then left in the "RESULT" module so that information commands can be called if desired. The program "GIFTSRUN" is executed in the following manner: PER
GIFTSRUN;LD01;PROM.SRC;PUSH.SRC.

The deflections and stresses obtained from the two loadings are contained and discussed in Appendix D. They were obtained by executing the INFON and INFST commands. The dynamical analysis of the capture system was done in a similar fashion using the program "DYNESRUN".

The crew/capture system on board the capture system presents a unique problem. In order to provide the degree of maneuverability necessary to safely intercept the moon ore modules, a number of thrusters must be attached to the tips of the A-beams of the structure as well as having a main propulsion engine located at the base of the structure. By developing a main propulsion system consisting of multiple, uni-directional engines, the majority of the maneuvers required of the structure may be implemented via the use of this system alone. To fine point the capture system and make slight adjustments prior to and during the interception of the moon ore, small hydrazine thrusters will be mounted in accordance with the three axes coordinate system on each A-beam. The main propulsion system will be powered by a fuel mixture of nitrogen and oxygen and can be refueled by depleting the reserve fuel left on board the shuttle and shuttle derived vehicles' main booster engines. Since the smaller hydrazine thrusters will use only a fraction of the fuel

used by the main engines, they may be refueled during a resupply mission at the same time as the thruster system on the habitat end manufacturing plant are refueled.

The attitude and control system of the capture system will be directed via the habitat facility. Information regarding the position of the capture system relative to the habitat and the incoming ore will be transmitted from the habitat communications center to both the ground facility and the capture structure. Horizon sensors, sun sensors, and star sensors will be mounted on the control enclosure of the capture system as well as inertial moment units that will correlate their data to the positioning information provided by the habitat. In case of a communications failure in the habitat, the capture structure will have the ability to transmit directly to the ground stations for assistance. The attitude control, however, will be provided by the propulsion system and the main purpose will be to align the capture system with the inbound ore.

Thermal control systems for the capture system will consist of both passive and active units. The cold gas thrusters and associated mechanisms located on the structures A-beams will be protected with entirely passive systems. The command and control facility onboard the capture system requires both active and passive thermal systems to ensure the proper working environments for both the crew and the computers. Active systems will consist of individual cooling and heating elements on computers and communication system. Between each component and on the exterior of the command structure will be insulation and thermal shielding to prevent excessive thermal expansion of the structure and possible system failure.

The manufacturing facility is an integrated system that may be

divided into three separate components, shown in Fig. 7, consisting of the refinery, the habitat, and the power system. The refinery section of the manufacturing plant will be capable of processing moon ore, allowing casting, forming, and final fabrication of the products as illustrated in Fig. 8 and 9.² After the final fabrication is complete, products from the facility are off loaded and stored in storage bin attached to the outside of the work area. At the end of the refinery is located an interfacing section that attaches the habitat to the refinery and allows either access or isolation of the two spaces in case of emergency. The interface section will include a sealed vacuum chamber for entry to and from the space environment as well as emergency rescue equipment and communication facilities.

The habitat structure will be made up of ten common modules that will be initially combined into four groups of two each. Each group will be launched separately and, once in orbit, the modules will be combined to provide crew quarters, a galley, medical and exercise areas, and a command center.³ The modules will be connected side by side, each with their own life support systems to include environmental control and thermal protection. To enable the facility to be easily resupplied, a docking area will be attached to the end of the habitat structure opposite the refinery. The docking area will have the ability to dock a shuttle directly to the habitat and will include a work area with equipment to work in the space environment. An illustration is shown on the following page.

In the design of the manufacturing facility and habitat, there were a number of different subsystems that required additional attention. Some of the subsystems included the power system, thermal protection, attitude

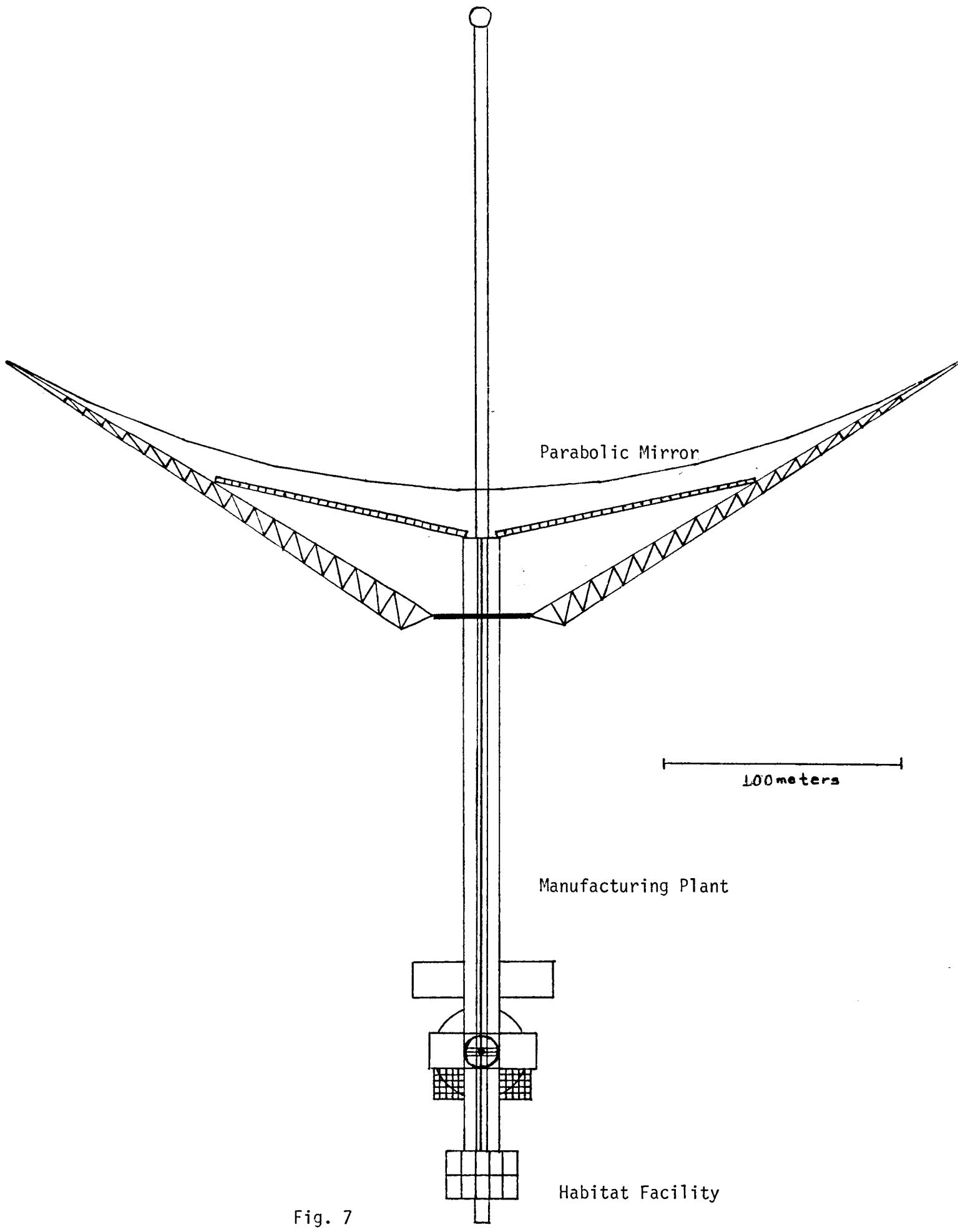
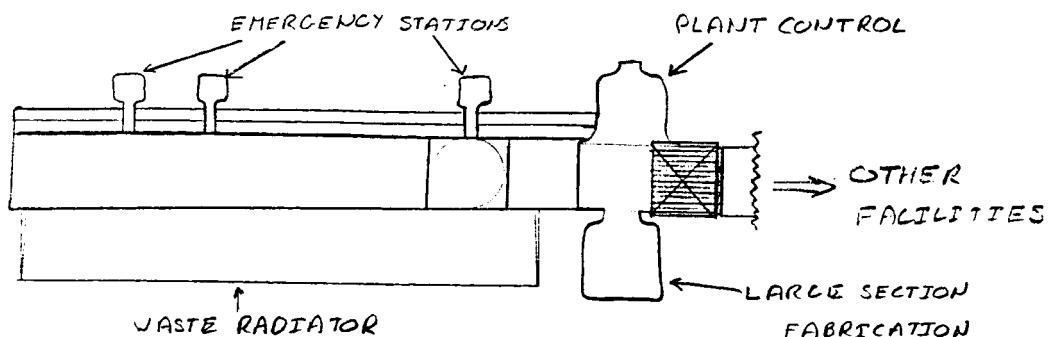


Fig. 7

Manufacturing Facility

VIEW I



VIEW II

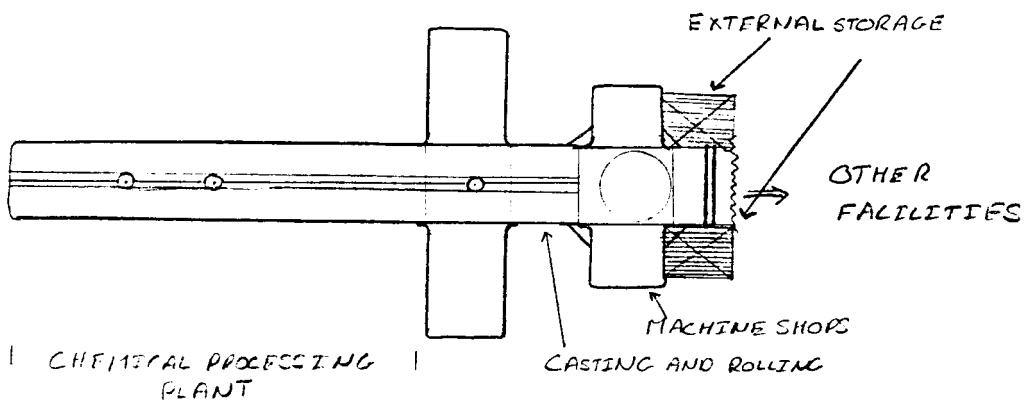


Fig. 8

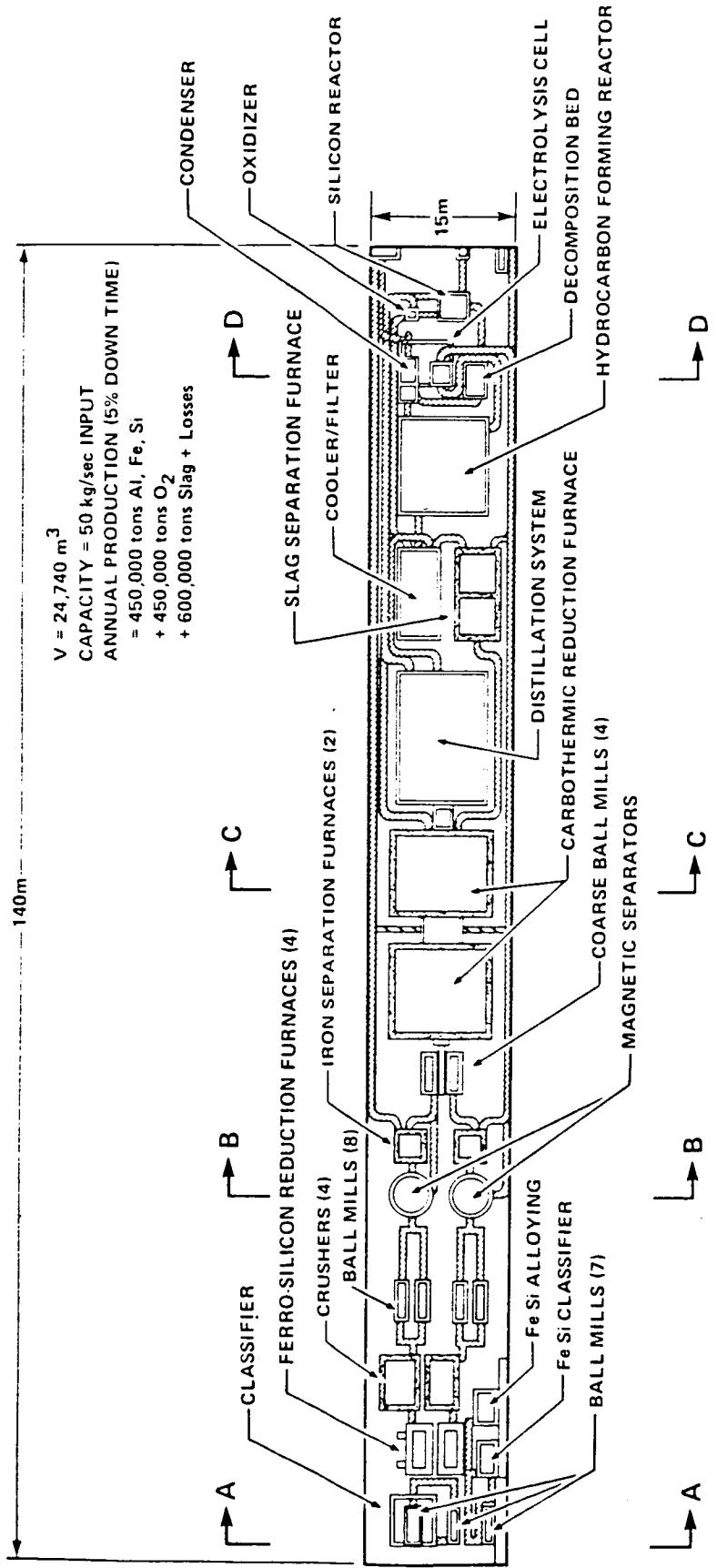


Fig. 9

Lateral-view schematic of the chemical process plant concept.

reference and control, orbital maintenance, and communications and data systems. The power system needed for the facility is largely dependent upon the power required to operate the refinery and associated working spaces. Approximately 56 Megawatts² are required to operate the manufacturing facility, including the habitat and a safety factor of about 1.1, the total energy requirements amount to 62 Megawatts. Conventional power supplies, such as a solar array or battery units, would have to be too large to be practical in order to supply the necessary power. A nuclear power plant would be able to supply the required amount of power but the hazards of placing a nuclear power plant in a LEO erases any feasibility. An advanced power system based on a parabolic mirror that focuses the solar energy on a focal point and thus concentrating the energy is the best prospect. As the solar energy is focused on the boiler, the energy is converted into thermal energy. The water, in turn, is pumped directly into the refinery, generating the required amount of power.

The thermal control system will actually consist of two separate systems, one system for the manufacturing plant and another one for the habitat. The manufacturing facility will have a thermal control system consisting primarily of a heat pipe radiator³ and passive systems such as insulation and shielding. Thermal control requirements for the habitat do not necessitate as large a system as that for the manufacturing plant. However, the habitat's system will consist of both active and passive thermal control devices. An active system may consist of a smaller version of the heat pipe radiator used on the manufacturing facility. In this way, the similarities of the two systems make it easier to maintain both systems and effect repairs. Also, slag from the refining process may be used in combination with the empty shuttle booster rockets for

both insulation and solar radiation protection of the habitat facility.

Attitude Reference and Control systems for the structure will be of a magnitude that has not been used in space to date. Since the size of the parabolic mirror greatly effects the amount of aerodynamic drag and moments acting on the facility, the control system must be powerful enough to maintain the proper attitude. With the use of dual sets of sun sensors, horizon sensors, and star sensors, the facility will be able to monitor a very exact attitude. The attitude control system will consist of cold gas hydrazine thrusters strategically located about the structure to enable motion in all directions of the three axes coordinate system. These thrusters, with the guidance from the attitude reference sensors and additional assistance from multiple Global Positioning Satellite (GPS) receivers, will be able to control the attitude to within 0.5 degrees.

The orbital maintenance system will be directly tied into the attitude control system. Since there is no need to provide an onboard propulsion system in order to boost the facility to final orbit, there will be no main propulsion engines. Instead, the array of thrusters required for attitude control will be complemented by a number of larger thrusters that will feed off the same hydrazine fuel storage. The larger thrusters will provide the necessary power to maintain the proper orbit.

Communications and data systems for the facility will include both ground-based and space-based networks. Located inside the habitat facility itself will be the main computer and communications network that will continuously record and transmit all orbital and attitude reference data automatically. Also located in the habitat will be an audio network and television system capable of voice and video communications from numerous positions throughout the facility. The manufacturing plant will house a separate communications network which will provide a

backups to the main network as well as supplying the main command center an information on the refinery and associated processes. All transmissions will be relayed, either directly or via communications satellites, to earth based stations located at White Sands.

After both the manufacturing facility and capture system are designed, each subsystem must be tested and interfaced with one another until the entire system can be integrated together. Once the manufacturing facility is built, it must be broken down into specific sections and arranged for easy assembly in space. In the case of the large parabolic mirror and the support beams of the capture system, each piece will be prefabricated prior to launch to enable quick and simple construction.

The initial in-space building phase of the system will begin with the launch and construction of the common modules, thus providing an initial space habitat to work from. The ten generic modules will be divided among two shuttle derived vehicle (SDV) payloads and one basic shuttle payload. The SDV used will be a heavy-lift launch vehicle using a side-mounted configuration similar to the one shown on the following page. In this configuration, with larger side mounted booster rockets, four common habitat modules may be launched in each SDV. The shuttle itself will carry two more modules to LEO. Completion of this initial phase will have the ten module habitat constructed in LEO within a two month time period. In the following two months, one SDV and one shuttle launch will be made. The SDV launch will carry the habitat-manufacturing plant interface structure as its payload. The shuttle payload will be the docking structure to be attached to the end of the habitat.

The manufacturing facility will be broken into ten SDV launches,

very much the same as the SDV used in the previous phase. These launches will be spaced out evenly over a period of ten months, after which there will be a two month period of no scheduled launches. During this period the refinery and habitat will be joined. The next phase is the launching of the materials for the power system. For ten months there will be one launch per month of materials to assemble the parabolic mirror and associated equipment.

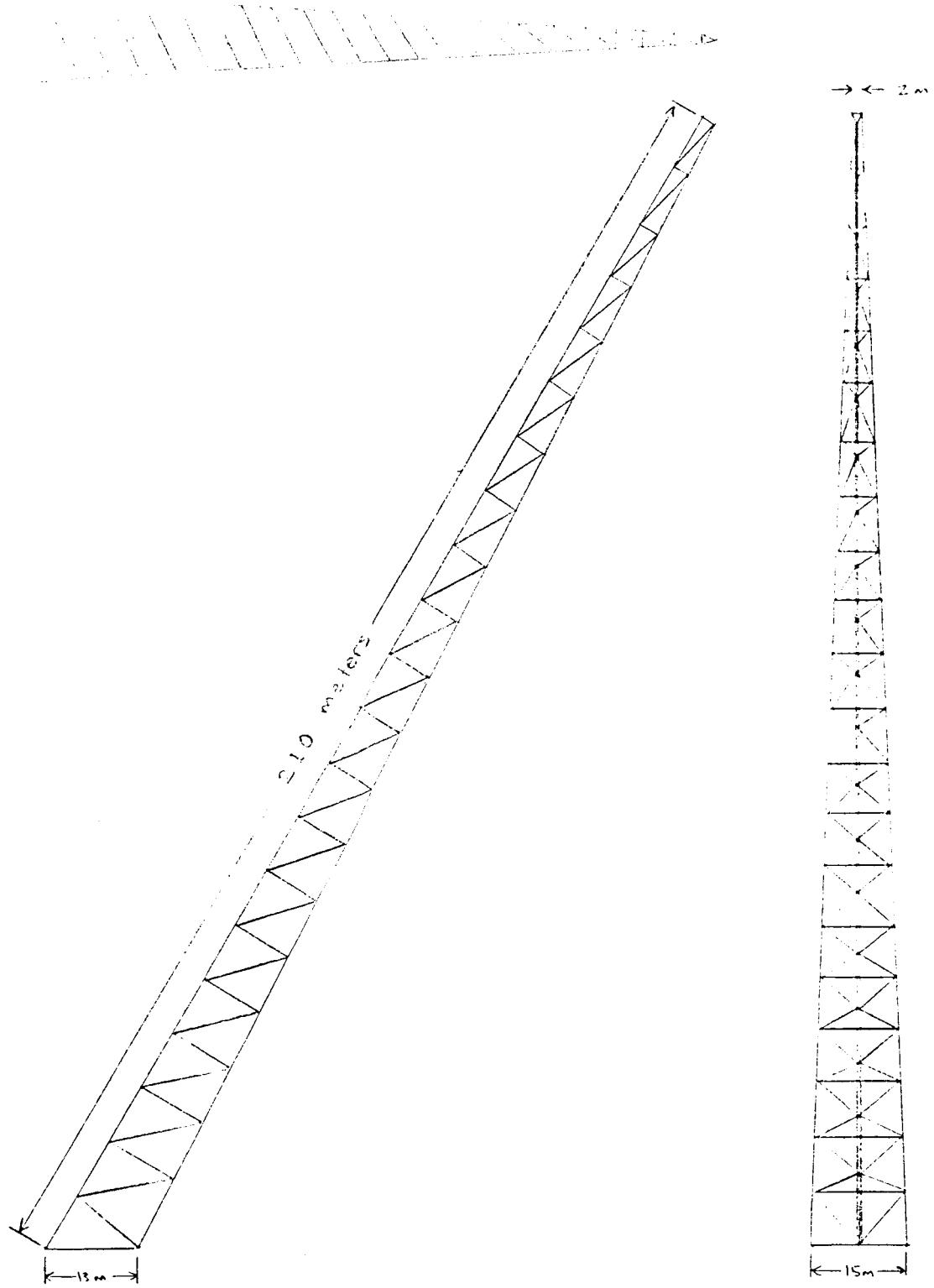
Once the manufacturing facility and habitat are located in LEO and are powered up, the building of the capture system may begin. The initial launch of the capture system components will occur three months after the completed refinery is in orbit. This phase will consist of one SDV and one shuttle launch. These first two launches will contain the command and control center and the lower frame of the capture system. Four additional SDV launches, spread over a four month period, will put into orbit the remaining support structures and the netting for the capture system. Once assembled, the capture system will power up, test systems, and begin communications with the habitat.

APPENDIXES

- A.....A-BEAM STRUCTURE
- B.....ALUMINUM ALLOY CHARACTERISTICS
- C.....C-BEAM STRUCTURE
- D.....GIFTS PROGRAMMING AND RESULTS

Appendix A

The A-beam



APPENDIX A

Appendix B

The Al Alloy

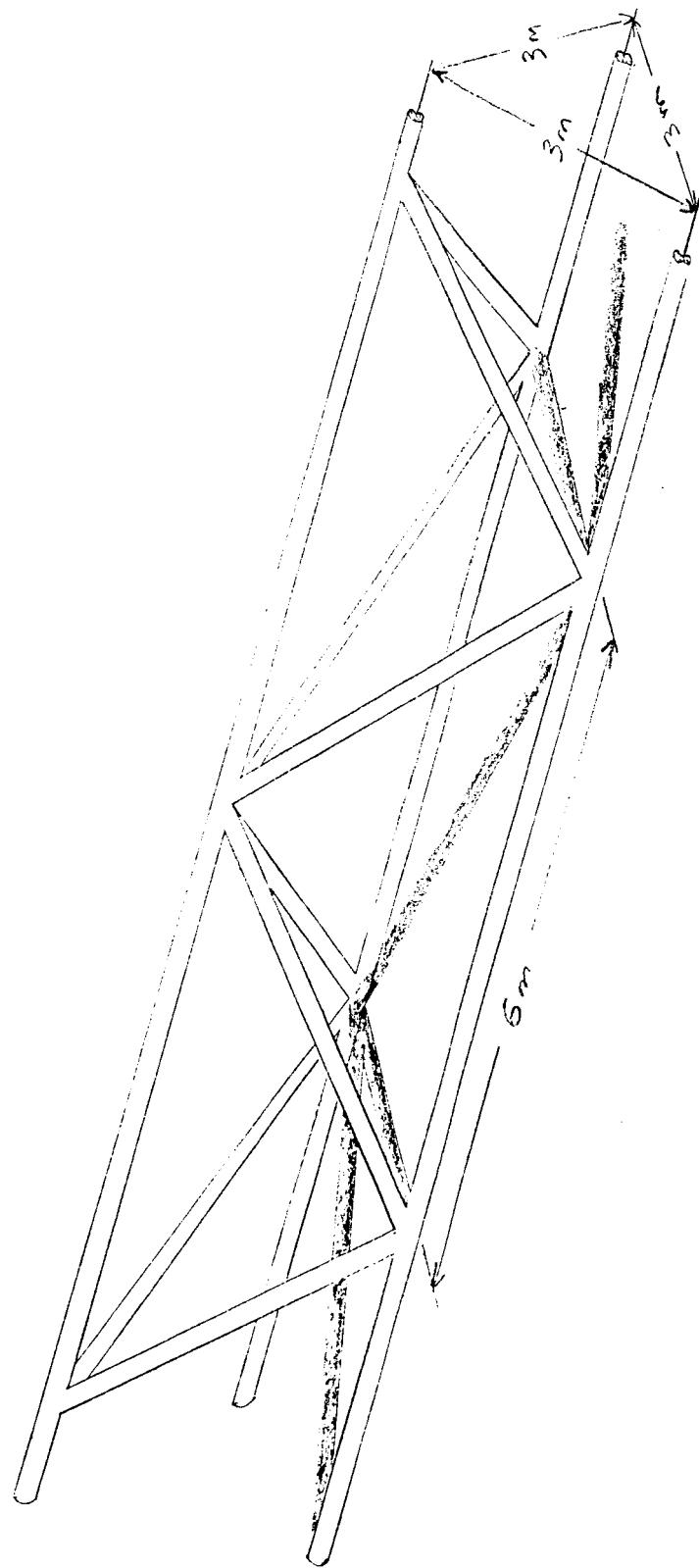
ALLOY 2014-T6 (4.4% Cu)

	<u>U.S. CUSTOMARY UNITS</u>	<u>SI UNITS</u>
Specific Weight	.101 lb/in ³	2800 kg/m ³
Ultimate Strength		
tension	70 ksi	480 MPa
compression	70 ksi	480 MPa
shear	42 ksi	290 MPa
Yield Strength		
tension	60 ksi	410 MPa
shear	32 ksi	220 MPa
Modulus of Elasticity	10.6×10^6 psi	72 GPa
Modulus of Rigidity	3.9×10^6 psi	27 GPa

APPENDIX B

Appendix C

The C-beam



APPENDIX C

Appendix D

GIFTS Data, Programs & Results

The job name CAS1 denotes the loading case for the off center hit in the upper half of the structure. An example of this is on the previous page. The first listing is that of the deflections. Note that the maximum deflection occurs at node #125,(dimensions are in feet). The second listing is that of the stresses in each member. Note that the maximum stresses occur in elements #175 and #181. They are 17.78 ksi and -18.14 ksi, respectfully. This leaves a safety factor of approximately 3 (max allowable= 60 ksi).

For the lower part of the structure, the case loading is denoted as CAS2. The maximum deflection (at node #125) is .9411 feet in the x-direction and .7107 feet in the z-direction. The way the beam was modeled, these are the only two directions of concern (unless, ofcourse, failure occurs). Again, the maximum stresses occur in element #'s 175 and 181. The safety factor in this beam is well over 3 and actually closer to 5.

LOADING CASE 1

19:50:24 10/10/96

14/28/96 19:50:24

THX THY

14/28/96 19:50:24

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32	-1.143E+00	5.995E-01	-9.036E-03		
33	-1.202E+00	8.064E-01	-9.517E-02		
34	-1.213E+00	7.143E-01	-6.303E-02		
35	-1.325E+00	7.143E-01	-2.052E-02		
36	-1.344E+00	8.259E-01	-9.871E-02		
37	-1.353E+00	7.414E-01	-3.557E-02		
38	-1.430E+00	7.414E-01	-3.201E-02		
39	-1.477E+00	8.494E-01	-1.012E-01		
40	-1.472E+00	7.181E-01	-3.919E-02		
41	-1.600E+00	7.181E-01	-4.567E-02		
42	-1.597E+00	8.421E-01	-1.025E-01		
43	-1.589E+00	7.371E-01	-4.172E-02		
44	-1.586E+00	7.371E-01	-6.924E-02		
45	-1.702E+00	8.124E-01	-1.028E-01		
46	-1.571E+00	8.372E-01	-4.534E-02		
47	-1.515E+00	8.372E-01	-7.515E-02		
48	-1.127E+00	7.332E-01	-1.020E-01		
49	-1.171E+00	5.795E-01	-4.786E-02		

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE	14/28/96	LOADING CASE	1	19:50:44	PAGE
NP	U	V	W	THX	THY
31	-1.064E+00	6.994E-01	-2.940E-02		
32	-1.143E+00	5.995E-01	-9.036E-03		
33	-1.202E+00	8.064E-01	-9.517E-02		
34	-1.213E+00	7.143E-01	-6.303E-02		
35	-1.325E+00	7.143E-01	-2.052E-02		
36	-1.344E+00	8.259E-01	-9.871E-02		
37	-1.353E+00	7.414E-01	-3.557E-02		
38	-1.430E+00	7.414E-01	-3.201E-02		
39	-1.477E+00	8.494E-01	-1.012E-01		
40	-1.472E+00	7.181E-01	-3.919E-02		
41	-1.600E+00	7.181E-01	-4.567E-02		
42	-1.597E+00	8.421E-01	-1.025E-01		
43	-1.589E+00	7.371E-01	-4.172E-02		
44	-1.586E+00	7.371E-01	-6.924E-02		
45	-1.702E+00	8.124E-01	-1.028E-01		
46	-1.571E+00	8.372E-01	-4.534E-02		
47	-1.515E+00	8.372E-01	-7.515E-02		
48	-1.127E+00	7.332E-01	-1.020E-01		
49	-1.171E+00	5.795E-01	-4.786E-02		

51	-1.870E+00	1.193E+01	1.18012E+01
52	-1.707E+00	1.154E+01	-5.148E+02
53	-1.534E+00	1.126E+01	-1.167E+01
54	-1.361E+00	1.098E+01	-4.105E+02
55	-1.188E+00	1.070E+01	-5.397E+02
56	-1.015E+00	1.042E+01	-1.270E+01
57	-8.422E+00	1.014E+01	-1.170E+01
58	-6.729E+00	1.010E+01	-5.097E+02
59	-5.036E+00	1.006E+01	-1.000E+01
60	-3.343E+00	1.002E+01	0.000E+01

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE	04/28/94	LOADING CASE	1	19:50:53	PAGE
NP	U	V	W	THX	THY
61	-1.200E+01	1.000E+01	0.000E+01		
62	-1.100E+01	1.000E+01	0.000E+01		
63	-1.000E+01	1.000E+01	-1.048E+01		
64	-1.214E+00	-4.908E+01	3.497E+01		
65	-1.442E+00	2.362E+01	-6.303E+01		
66	-1.072E+01	2.352E+01	-2.857E+01		
67	-1.650E+00	-4.753E+01	1.491E+02		
68	-1.710E+00	3.384E+01	-5.864E+01		
69	-1.765E+00	3.391E+01	-4.143E+01		
70	-1.875E+00	-4.584E+01	1.644E+01		
71	-2.133E+00	2.423E+01	-8.084E+01		
72	-1.574E+00	3.415E+01	-4.944E+01		
73	-2.119E+00	-4.425E+01	-4.597E+01		
74	-3.426E+00	2.442E+01	-8.798E+01		
75	-1.874E+00	2.450E+01	-5.282E+01		
76	-2.007E+00	-4.227E+01	-6.016E+01		
77	-1.808E+00	2.463E+01	-8.027E+01		
78	-1.955E+00	2.474E+01	-5.188E+01		
79	-1.823E+00	-4.027E+01	-6.919E+01		
80	-1.673E+00	2.486E+01	-8.752E+01		
81	-1.754E+00	2.493E+01	-4.687E+01		
82	-1.248E+00	-8.779E+01	-7.024E+01		
83	-3.422E+00	2.485E+01	-8.103E+01		
84	-1.855E+00	2.490E+01	-3.843E+01		
85	-2.092E+00	-8.481E+01	-6.697E+01		
86	-2.513E+00	2.462E+01	-7.087E+01		
87	-1.677E+00	2.431E+01	-2.718E+01		
88	-1.868E+00	-8.118E+01	-5.859E+01		
89	-2.297E+00	2.463E+01	-5.717E+01		
90	-1.438E+00	2.440E+01	-1.399E+01		

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE	04/28/94	LOADING CASE	1	19:51:12	PAGE
NP	U	V	W	THX	THY
91	-1.593E+00	-2.668E+01	-4.588E+01		
92	-2.016E+00	2.472E+01	-4.102E+01		
93	-1.150E+00	2.651E+01	5.012E+03		
94	-1.297E+00	-2.141E+01	-3.020E+01		
95	-1.681E+00	2.497E+01	-2.345E+01		
96	-9.079E+01	2.520E+01	1.069E+01		
97	-1.033E+00	-1.712E+01	-1.656E+01		
98	-1.387E+00	2.548E+01	-1.052E+01		
99	-6.517E+01	2.576E+01	2.234E+01		
100	-7.692E+01	-1.304E+01	-2.575E+02		
101	-1.090E+00	2.575E+01	2.896E+02		
102	-3.947E+01	2.636E+01	3.404E+01		
103	-5.014E+01	-8.962E+02	1.125E+01		
104	-7.324E+01	2.534E+01	1.637E+01		
105	-1.054E+01	2.710E+01	4.578E+01		
106	-2.226E+01	-4.815E+02	2.522E+01		
107	-4.512E+01	2.700E+01	2.988E+01		

111	1.125E+01	1.113E+01
112	1.125E+01	1.079E+01
113	1.125E+01	1.045E+01
114	1.125E+01	1.010E+01
115	1.125E+01	9.745E+00
116	1.125E+01	9.390E+00
117	1.125E+01	9.035E+00
118	1.125E+01	8.680E+00
119	1.125E+01	8.325E+00
120	1.125E+01	7.970E+00
121	1.125E+01	7.615E+00
122	1.125E+01	7.260E+00
123	1.125E+01	6.905E+00
124	1.125E+01	6.550E+00
125	1.125E+01	6.195E+00
126	1.125E+01	5.840E+00
127	1.125E+01	5.485E+00
128	1.125E+01	5.130E+00
129	1.125E+01	4.775E+00
130	1.125E+01	4.420E+00
131	1.125E+01	4.065E+00
132	1.125E+01	3.710E+00
133	1.125E+01	3.355E+00
134	1.125E+01	2.990E+00
135	1.125E+01	2.635E+00
136	1.125E+01	2.270E+00
137	1.125E+01	1.905E+00
138	1.125E+01	1.540E+00
139	1.125E+01	1.175E+00
140	1.125E+01	8.095E-01
141	1.125E+01	4.415E-01
142	1.125E+01	6.835E-02
143	1.125E+01	1.125E-02
144	1.125E+01	1.565E-03
145	1.125E+01	2.005E-04
146	1.125E+01	2.445E-05
147	1.125E+01	2.885E-06
148	1.125E+01	3.325E-07
149	1.125E+01	3.765E-08
150	1.125E+01	4.205E-09
151	1.125E+01	4.645E-10
152	1.125E+01	5.085E-11
153	1.125E+01	5.525E-12
154	1.125E+01	5.965E-13
155	1.125E+01	6.405E-14
156	1.125E+01	6.845E-15
157	1.125E+01	7.285E-16
158	1.125E+01	7.725E-17
159	1.125E+01	8.165E-18
160	1.125E+01	8.605E-19
161	1.125E+01	9.045E-20
162	1.125E+01	9.485E-21
163	1.125E+01	9.925E-22
164	1.125E+01	1.036E-22
165	1.125E+01	1.080E-23
166	1.125E+01	1.124E-24
167	1.125E+01	1.168E-25
168	1.125E+01	1.212E-26
169	1.125E+01	1.256E-27
170	1.125E+01	1.300E-28
171	1.125E+01	1.344E-29
172	1.125E+01	1.388E-30
173	1.125E+01	1.432E-31
174	1.125E+01	1.476E-32
175	1.125E+01	1.520E-33
176	1.125E+01	1.564E-34
177	1.125E+01	1.608E-35
178	1.125E+01	1.652E-36
179	1.125E+01	1.696E-37
180	1.125E+01	1.740E-38
181	1.125E+01	1.784E-39
182	1.125E+01	1.828E-40
183	1.125E+01	1.872E-41
184	1.125E+01	1.916E-42
185	1.125E+01	1.960E-43
186	1.125E+01	2.004E-44
187	1.125E+01	2.048E-45
188	1.125E+01	2.092E-46
189	1.125E+01	2.136E-47
190	1.125E+01	2.180E-48
191	1.125E+01	2.224E-49
192	1.125E+01	2.268E-50
193	1.125E+01	2.312E-51
194	1.125E+01	2.356E-52
195	1.125E+01	2.400E-53
196	1.125E+01	2.444E-54
197	1.125E+01	2.488E-55
198	1.125E+01	2.532E-56
199	1.125E+01	2.576E-57
200	1.125E+01	2.620E-58
201	1.125E+01	2.664E-59
202	1.125E+01	2.708E-60
203	1.125E+01	2.752E-61
204	1.125E+01	2.796E-62
205	1.125E+01	2.840E-63
206	1.125E+01	2.884E-64
207	1.125E+01	2.928E-65
208	1.125E+01	2.972E-66
209	1.125E+01	3.016E-67
210	1.125E+01	3.060E-68
211	1.125E+01	3.104E-69
212	1.125E+01	3.148E-70
213	1.125E+01	3.192E-71
214	1.125E+01	3.236E-72
215	1.125E+01	3.280E-73
216	1.125E+01	3.324E-74
217	1.125E+01	3.368E-75
218	1.125E+01	3.412E-76
219	1.125E+01	3.456E-77
220	1.125E+01	3.500E-78
221	1.125E+01	3.544E-79
222	1.125E+01	3.588E-80
223	1.125E+01	3.632E-81
224	1.125E+01	3.676E-82
225	1.125E+01	3.720E-83
226	1.125E+01	3.764E-84
227	1.125E+01	3.808E-85
228	1.125E+01	3.852E-86
229	1.125E+01	3.896E-87
230	1.125E+01	3.940E-88
231	1.125E+01	3.984E-89
232	1.125E+01	4.028E-90
233	1.125E+01	4.072E-91
234	1.125E+01	4.116E-92
235	1.125E+01	4.160E-93
236	1.125E+01	4.204E-94
237	1.125E+01	4.248E-95
238	1.125E+01	4.292E-96
239	1.125E+01	4.336E-97
240	1.125E+01	4.380E-98
241	1.125E+01	4.424E-99
242	1.125E+01	4.468E-100
243	1.125E+01	4.512E-101
244	1.125E+01	4.556E-102
245	1.125E+01	4.600E-103
246	1.125E+01	4.644E-104
247	1.125E+01	4.688E-105
248	1.125E+01	4.732E-106
249	1.125E+01	4.776E-107
250	1.125E+01	4.820E-108
251	1.125E+01	4.864E-109
252	1.125E+01	4.908E-110
253	1.125E+01	4.952E-111
254	1.125E+01	5.000E-112
255	1.125E+01	5.048E-113
256	1.125E+01	5.096E-114
257	1.125E+01	5.144E-115
258	1.125E+01	5.192E-116
259	1.125E+01	5.240E-117
260	1.125E+01	5.288E-118
261	1.125E+01	5.336E-119
262	1.125E+01	5.384E-120
263	1.125E+01	5.432E-121
264	1.125E+01	5.480E-122
265	1.125E+01	5.528E-123
266	1.125E+01	5.576E-124
267	1.125E+01	5.624E-125
268	1.125E+01	5.672E-126
269	1.125E+01	5.720E-127
270	1.125E+01	5.768E-128
271	1.125E+01	5.816E-129
272	1.125E+01	5.864E-130
273	1.125E+01	5.912E-131
274	1.125E+01	5.960E-132
275	1.125E+01	6.008E-133
276	1.125E+01	6.056E-134
277	1.125E+01	6.104E-135
278	1.125E+01	6.152E-136
279	1.125E+01	6.199E-137
280	1.125E+01	6.247E-138
281	1.125E+01	6.295E-139
282	1.125E+01	6.343E-140
283	1.125E+01	6.391E-141
284	1.125E+01	6.439E-142
285	1.125E+01	6.487E-143
286	1.125E+01	6.535E-144
287	1.125E+01	6.583E-145
288	1.125E+01	6.631E-146
289	1.125E+01	6.679E-147
290	1.125E+01	6.727E-148
291	1.125E+01	6.775E-149
292	1.125E+01	6.823E-150
293	1.125E+01	6.871E-151
294	1.125E+01	6.919E-152
295	1.125E+01	6.967E-153
296	1.125E+01	7.015E-154
297	1.125E+01	7.063E-155
298	1.125E+01	7.111E-156
299	1.125E+01	7.159E-157
300	1.125E+01	7.207E-158
301	1.125E+01	7.255E-159
302	1.125E+01	7.303E-160
303	1.125E+01	7.351E-161
304	1.125E+01	7.399E-162
305	1.125E+01	7.447E-163
306	1.125E+01	7.495E-164
307	1.125E+01	7.543E-165
308	1.125E+01	7.591E-166
309	1.125E+01	7.639E-167
310	1.125E+01	7.687E-168
311	1.125E+01	7.735E-169
312	1.125E+01	7.783E-170
313	1.125E+01	7.831E-171
314	1.125E+01	7.879E-172
315	1.125E+01	7.927E-173
316	1.125E+01	7.975E-174
317	1.125E+01	8.023E-175
318	1.125E+01	8.071E-176
319	1.125E+01	8.119E-177
320	1.125E+01	8.167E-178
321	1.125E+01	8.215E-179
322	1.125E+01	8.263E-180
323	1.125E+01	8.311E-181
324	1.125E+01	8.359E-182
325	1.125E+01	8.407E-183
326	1.125E+01	8.455E-184
327	1.125E+01	8.503E-185
328	1.125E+01	8.551E-186
329	1.125E+01	8.599E-187
330	1.125E+01	8.647E-188
331	1.125E+01	8.695E-189
332	1.125E+01	8.743E-190
333	1.125E+01	8.791E-191
334	1.125E+01	8.839E-192
335	1.125E+01	8.887E-193
336	1.125E+01	8.935E-194
337	1.125E+01	8.983E-195
338	1.125E+01	9.031E-196
339	1.125E+01	9.079E-197
340	1.125E+01	9.127E-198
341	1.125E+01	9.175E-199
342	1.125E+01	9.223E-200
343	1.125E+01	9.271E-201
344	1.125E+01	9.319E-202
345	1.125E+01	9.367E-203
346	1.125E+01	9.415E-204
347	1.125E+01	9.463E-205
348	1.125E+01	9.511E-206
349	1.125E+01	9.559E-207
350	1.125E+01	9.607E-208
351	1.125E+01	9.655E-209
352	1.125E+01	9.703E-210
353	1.125E+01	9.751E-211
354	1.125E+01	9.799E-212
355	1.125E+01	9.847E-213
356	1.125E+01	9.895E-214
357	1.125E+01	9.943E-215
358	1.125E+01	1.0000E+00

143:04:31

14/23/86

LOADING CASE 1

19:51:31

PAGE 7

THY

TIME

144:04:31

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LOADING CASE 1

19:51:31

PAGE 8

THY

TIME

145:04:31

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LOADING CASE 1

19:51:31

PAGE 9

THY

TIME

146:04:31

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LOADING CASE 1

19:51:31

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THY

TIME

147:04:31

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LOADING CASE 1

19:51:31

JOB:CASE1 04/28/86

LOADING CASE 1

STRESSES -- MIDDLE

31 6.462E+03
32 1.257E+04
33 2.890E+03
34 -1.449E+03
35 4.530E+03
36 1.258E+04
37 2.890E+03
38 -1.449E+03
39 6.441E+03
40 1.643E+04
41 2.889E+03
42 -1.450E+03
43 4.508E+03
44 1.644E+04
45 2.887E+03
46 -1.450E+03
47 6.431E+03
48 2.887E+04
49 2.887E+03
50 -1.451E+03
51 1.431E+03
52 2.887E+04

TO CONTINUE, STRIKE RETURN; TO STOP, TYPE 'S'?

JOB:CASE1

04/28/86

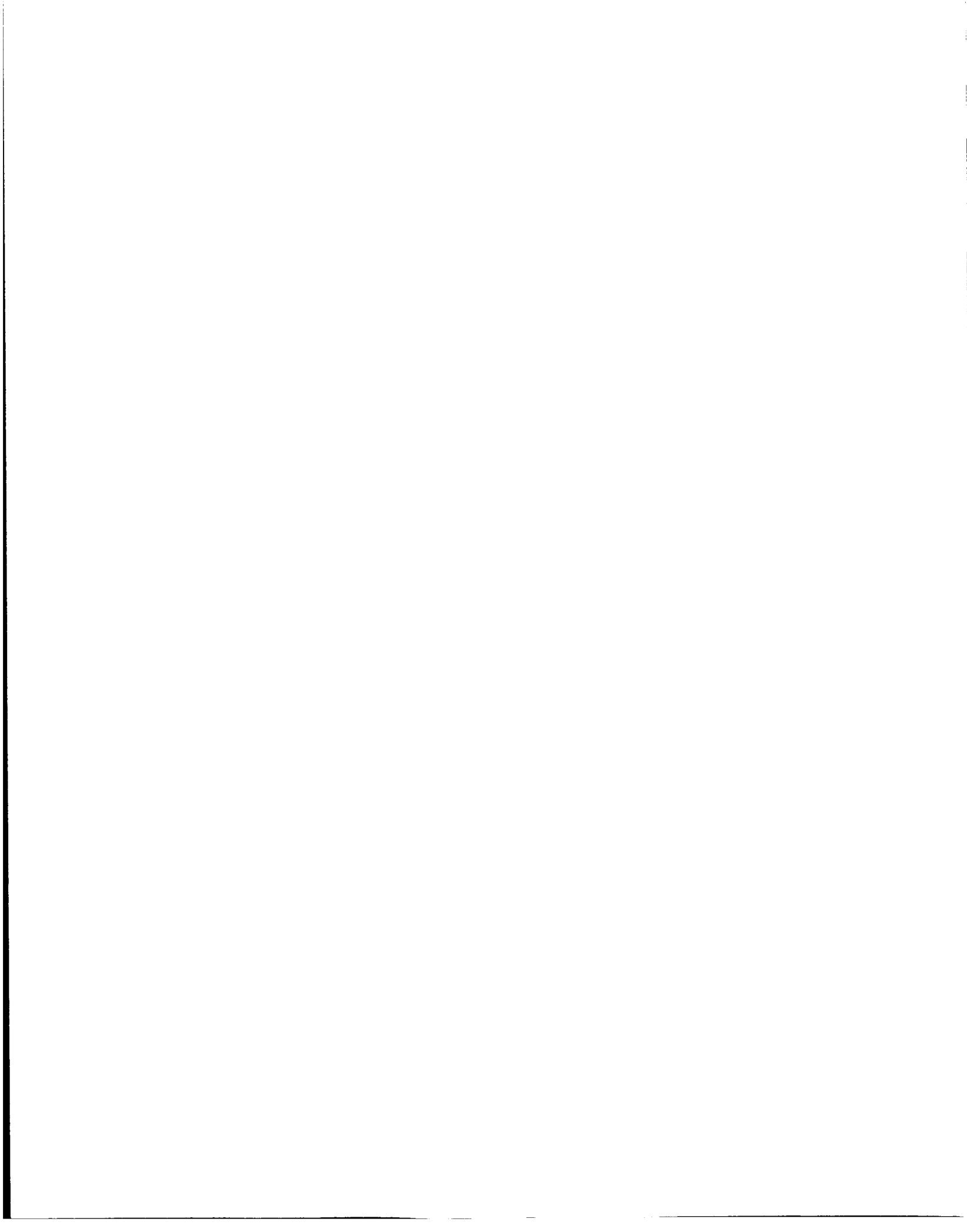
LOADING CASE 1

19:52:38

PAGE 2

ME

31 6.462E+03
32 1.257E+04
33 2.890E+03
34 -1.449E+03
35 4.530E+03
36 1.258E+04
37 2.890E+03
38 -1.449E+03
39 6.441E+03
40 1.643E+04
41 2.889E+03
42 -1.450E+03
43 4.508E+03
44 1.644E+04
45 2.887E+03
46 -1.450E+03
47 6.431E+03
48 2.887E+04
49 2.887E+03
50 -1.451E+03
51 1.431E+03
52 2.887E+04



TO CONTINUE, STRIKE <RETURN>; TO STOP, TYPE '&S'?

JOB:CASE1 04/28/86 LOADING CASE 1 19:52:51 PAGE 3
STRESSES -- MIDDLE

91	1.168E+03
92	6.250E+02
93	4.033E+02
94	2.419E+02
95	5.032E+02
96	-1.409E+03
97	-3.320E+03
98	-5.229E+03
99	-7.136E+03
100	-1.357E+03
101	-2.146E+03
102	-2.147E+03
103	-2.148E+03
104	-2.154E+03
105	-2.157E+03
106	-2.157E+03
107	-1.476E+04
108	-1.148E+04
109	-1.148E+04
110	-1.147E+04

TO CONTINUE, STRIKE <RETURN>; TO STOP, TYPE '&S'?

JOB:CASE1 04/28/86 LOADING CASE 1 19:53:02 PAGE 4
NE STRESSES -- MIDDLE

91	8.168E+03
92	6.250E+03
93	4.033E+03
94	2.419E+03
95	5.032E+02
96	-1.409E+03
97	-3.320E+03
98	-5.229E+03
99	-7.136E+03
100	-1.357E+03
101	-2.146E+03
102	-2.147E+03
103	-2.148E+03
104	-2.154E+03
105	-2.157E+03
106	-2.157E+03
107	-1.476E+04
108	-1.148E+04
109	-1.148E+04
110	-1.147E+04

151 -2.889E+03
152 1.450E+03
153 -2.889E+03
154 1.450E+03
155 -2.888E+03
156 1.450E+03
157 -2.888E+03
158 1.451E+03
159 -2.888E+03
160 1.452E+03
161 -2.887E+03
162 1.453E+03
163 -2.887E+03
164 1.454E+03
165 -2.886E+03
166 1.454E+03
167 -2.886E+03
168 1.455E+03

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE1 04/26/86
NE
151 -2.889E+03
152 1.450E+03
153 -2.889E+03
154 1.450E+03
155 -2.888E+03
156 1.450E+03
157 -2.888E+03
158 1.451E+03
159 -2.888E+03
160 1.452E+03
161 -2.887E+03
162 1.453E+03
163 -2.887E+03
164 1.454E+03
165 -2.886E+03
166 1.454E+03
167 -2.886E+03
168 1.455E+03

04/26/86

LOADING CASE 1
STRESSES -- MIDDLE

19:53:14

PAGE 5

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE1 04/28/86
NE
151 -2.889E+03
152 1.450E+03
153 -2.889E+03
154 1.450E+03
155 -2.888E+03
156 1.450E+03
157 -2.888E+03
158 1.451E+03
159 -2.888E+03
160 1.452E+03
161 -2.887E+03
162 1.453E+03
163 -2.887E+03
164 1.454E+03
165 -2.886E+03
166 1.454E+03
167 -2.886E+03
168 1.455E+03

04/28/86

LOADING CASE 1
STRESSES -- MIDDLE

19:53:27

PAGE 6

1. *On the Nature of the Human Soul*, by Dr. J. H. Newman, M.A., F.R.S., &c. &c. &c.

11. 100% OF STRIKE & RETURN TO STOP, TYPE 134?

04/23/96

LOADING CASE 1
MIDSPAN - 11 MILES

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- 1 -

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

```

JOB:CASE1
NE
211 -8.3505E+03
212 -4.4675E+04
213 -3.8798E+03
214 -4.9544E+04
215 -3.0305E+03
216 -4.1138E+03
217 -8.3438E+03
218 -3.8235E+04
219 -4.2205E+03
220 -4.4413E+04
221 -3.2168E+03
222 -4.1714E+03
223 -9.1396E+03
224 -3.1114E+03
225 -7.4114E+03
226 -3.0583E+04

```

04/29/94

LOADING CASE 1
STRESSES -- MIDDLE

19:04 06/01

PAGE 3

241 -1.098E+00
242 -1.567E+00
243 -1.257E+00
244 -1.098E+00
245 -1.098E+00
246 -1.098E+00
247 -1.098E+00
248 -1.098E+00
249 -1.098E+00
250 -1.098E+00
251 -1.098E+00
252 -1.098E+00
253 -1.098E+00
254 -1.098E+00
255 -1.098E+00
256 -1.098E+00
257 -1.098E+00
258 -1.098E+00
259 -1.098E+00
260 -1.098E+00
261 -1.098E+00
262 -1.098E+00
263 -1.098E+00
264 -1.098E+00
265 -1.098E+00
266 -1.098E+00
267 -1.098E+00
268 -1.098E+00
269 -1.098E+00
270 -1.098E+00

TO CONTINUE, STRIKE 'RETURN': TO STOP, TYPE 'S'?

JOB:0481 04/28/86 LOADING CASE 1 19:54:56 PAGE 12
STRESSES -- MIDDLE

271 -1.432E+01
272 -2.621E+01
273 1.951E+01
274 -2.803E+02
275 -2.215E+00
276 -2.441E+00
277 -1.559E+01
278 -2.483E+01
279 1.987E+01
280 -2.317E+02
281 -2.457E+00
282 -2.464E+01
283 -2.453E+01
284 -2.363E+01

TO CONTINUE, STRIKE 'RETURN': TO STOP, TYPE 'S'?

JOB:0481 04/28/86 LOADING CASE 1 19:55:11 PAGE 13
NE STRESSES -- MIDDLE

271 -1.432E+01
272 -2.621E+01
273 1.951E+01
274 -2.803E+02
275 -2.215E+00
276 -2.441E+00
277 -1.559E+01
278 -2.483E+01
279 1.987E+01
280 -2.317E+02
281 -2.457E+00
282 -2.464E+01
283 -2.453E+01
284 -2.363E+01

331 1.5892E+04
332 4.0288E+02
333 -2.9658E+03
334 -1.2458E+03
335 6.8608E+02
336 8.7688E+00
337 1.1148E+01
338 2.1588E+02
339 1.5288E+01
340 1.7418E+01
341 1.15288E+02
342 1.4838E+01

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'Q'?

JOB:0481 04/28/86 STRESSES -- MIDDLE
331 -1.5892E+04
332 1.7144E+02
333 1.0938E+03
334 1.4124E+03
335 1.2098E+03
336 1.1208E+03
337 1.0728E+03
338 1.0328E+03
339 1.0028E+03
340 9.7228E+02
341 9.4228E+02
342 9.1228E+02
343 8.8228E+02
344 8.5228E+02
345 8.2228E+02
346 7.9228E+02
347 7.6228E+02
348 7.3228E+02
349 7.0228E+02
350 6.7228E+02
351 6.4228E+02
352 6.1228E+02
353 5.8228E+02
354 5.5228E+02
355 5.2228E+02
356 4.9228E+02
357 4.6228E+02
358 4.3228E+02
359 4.0228E+02
360 3.7228E+02
361 3.4228E+02
362 3.1228E+02
363 2.8228E+02
364 2.5228E+02
365 2.2228E+02
366 1.9228E+02
367 1.6228E+02
368 1.3228E+02
369 1.0228E+02
370 7.2228E+01
371 4.2228E+01
372 1.2228E+01
373 -1.2228E+01
374 -4.2228E+01
375 -7.2228E+01
376 -1.2228E+02
377 -1.5228E+02
378 -1.8228E+02
379 -2.1228E+02
380 -2.4228E+02

LOADING CASE 1
STRESSES -- MIDDLE

19:55:24

PAGE 11

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'Q'?

JOB:0481 04/28/86 STRESSES -- MIDDLE
NE
331 1.5892E+04
332 4.0288E+02
333 -2.9658E+03
334 -1.2458E+03
335 6.8608E+02
336 8.7688E+00
337 1.1148E+01
338 2.1588E+02
339 1.5288E+01
340 1.7418E+01
341 1.15288E+02
342 1.4838E+01

LOADING CASE 1
STRESSES -- MIDDLE

19:55:38

PAGE 12

10. 100% TIME, STRIKE & RETURNING TO STOP, TYPE 100%

00041-01 04/28/86 LOADING CASE 1 19:55:51 7468 11
1 E STREESSES -- MIDDLE

	4	6	8	10	12	14	16	18	20	22	24	26	28	30	32	34	36	38	40	42	44	46	48	50	52	54	56	58	60	62	64	66	68	70	72	74	76	78	80	82	84	86	88	90	92	94	96	98	100
--	---	---	---	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	----	-----

二〇一〇年

LOADING CASE 2

（五）在本办法施行前，已经完成登记的，应当在本办法施行后六个月内，向登记机关申请换发登记证书。

1. The following table gives the number of cases of smallpox reported in each of the 100 districts of the United States during the year 1881.

1990-1991 学年第一学期期中考试高二数学(文科)参考答案

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1. 100 100 100 100 100 100 100 100 100

REFERENCES AND NOTES

— 1 —

LINE	DATE	LOADING	CASE	TIME	TIME	PAGE
1	1.000E+01	0.000E+01	0.000E+01			
2	1.000E+01	0.000E+01	0.000E+01			
3	1.000E+01	0.000E+01	0.000E+01			
4	-3.120E-02	3.595E-03	-8.120E-04			
5	-5.310E-03	9.585E-03	9.015E-03			
6	-1.381E-11	2.274E-02	-9.966E-03			
7	-2.341E-02	3.871E-02	-1.898E-02			
8	-3.552E-02	2.871E-02	8.438E-03			
9	-5.533E-02	4.680E-02	-1.918E-02			
10	-6.932E-02	4.474E-02	-8.668E-03			
11	-9.346E-02	4.474E-02	6.970E-03			
12	-1.161E-01	7.125E-02	-2.745E-02			
13	-1.262E-01	6.676E-02	-4.890E-03			
14	-1.428E-01	6.676E-02	5.829E-03			
15	-1.791E-01	9.459E-02	-3.536E-02			
16	-1.905E-01	7.922E-02	-6.799E-03			
17	-2.144E-01	7.922E-02	2.197E-03			
18	-2.438E-01	1.157E-01	-4.231E-02			
19	-2.548E-01	9.325E-02	-8.166E-03			
20	-2.950E-01	9.825E-02	-9.076E-04			
21	-3.242E-01	1.038E-01	-4.852E-02			
22	-3.485E-01	1.023E-01	-1.021E-02			
23	-3.980E-01	1.023E-01	-5.508E-03			
24	-4.138E-01	1.442E-01	-5.397E-02			
25	-4.393E-01	1.1134E-01	-1.172E-02			
26	-4.755E-01	1.1154E-01	-9.770E-03			
27	-5.112E-01	1.531E-01	-5.867E-02			
28	-5.474E-01	1.041E-01	-1.091E-02			
29	-5.835E-01	1.1041E-01	-1.1553E-02			
30	-6.197E-01	1.551E-01	-7.405E-03			

100 : 0482 04/28/96 LOADING CASE 1 15:31:17 14:54
 100 : 0482 04/28/96 W THX THY
 1 1.325E+01 -1.356E+02
 2 -1.325E+01 1.325E+01 -2.138E+02
 3 -1.485E+01 1.485E+01 -3.582E+02
 4 -1.747E+01 7.475E+02 -1.789E+02
 5 -1.154E+01 7.495E+02 -2.846E+02
 6 -1.153E+01 1.264E+01 -3.827E+02
 7 -1.153E+01 1.563E+02 -1.968E+02
 8 -1.477E+01 1.563E+02 -3.1367E+02
 9 -1.152E+01 1.756E+02 -4.997E+02
 10 -1.623E+01 1.927E+03 -2.215E+02
 11 -1.623E+01 3.025E+03 -4.415E+02
 12 -1.623E+01 5.466E+02 -7.091E+02
 13 -1.623E+01 -3.692E+02 -2.408E+02
 14 -1.701E+00 -3.582E+02 -5.270E+02
 15 -1.003E+00 2.499E+03 -7.111E+02
 16 -1.282E+01 -1.155E+01 -2.668E+02
 17 -1.672E+00 -1.155E+01 -6.255E+02
 18 -1.255E+00 -6.613E+02 -7.056E+02
 19 -1.138E+00 -1.347E+01 -2.875E+02
 20 -1.117E+00 -1.347E+01 -7.246E+02
 21 -1.117E+00 -1.505E+01 -3.926E+02
 22 -1.118E+01 -2.963E+01 -3.148E+02
 23 -1.118E+01 -2.963E+01 -8.366E+02
 24 -1.117E+00 -1.517E+01 -6.722E+02
 25 -1.117E+00 -3.990E+01 -3.368E+02
 26 -1.118E+00 -3.988E+01 -3.494E+02
 27 -1.118E+00 -3.712E+01 -6.443E+02
 28 -1.117E+01 -4.627E+01 -3.368E+02
 29 -1.116E+00 -4.353E+01 -9.666E+02
 30 -1.000E+01 0.000E+01 0.000E+01

TO CONTINUE, STRIKE 1 RETURN; TO STOP, TYPE '1812'

100 : 0482 04/28/96 LOADING CASE 1 15:31:17 14:54
 100 : 0482 04/28/96 W THX THY
 1 0.000E+01 0.000E+01 0.000E+01
 2 0.000E+01 0.000E+01 0.000E+01
 3 -2.455E+01 9.361E+02 -6.568E+02
 4 -7.227E+01 -3.170E+01 2.017E+01
 5 -9.179E+01 9.759E+02 -4.481E+01
 6 -5.945E+01 4.234E+02 -1.810E+01
 7 -1.000E+00 -3.656E+01 -1.243E+02
 8 -3.857E+01 4.421E+02 -4.824E+01
 9 -8.637E+01 -1.040E+02 -2.628E+01
 10 -1.207E+00 -4.114E+01 -1.845E+01
 11 -1.262E+00 -8.543E+03 -5.592E+01
 12 -1.058E+00 -6.457E+02 -3.132E+01
 13 -1.344E+00 -4.681E+01 -3.150E+01
 14 -1.453E+00 -6.272E+02 -6.035E+01
 15 -1.190E+00 -1.201E+01 -3.338E+01
 16 -1.414E+00 -5.112E+01 -4.047E+01
 17 -1.569E+00 -1.182E+01 -6.165E+01
 18 -1.232E+00 -1.758E+01 -3.243E+01
 19 -1.423E+00 -5.563E+01 -4.548E+01
 20 -1.618E+00 -1.749E+01 -5.997E+01
 21 -1.220E+00 -2.345E+01 -2.932E+01
 22 -1.370E+00 -6.038E+01 -4.371E+01
 23 -1.381E+00 -2.327E+01 -5.554E+01
 24 -1.150E+00 -2.992E+01 -2.877E+01
 25 -1.151E+01 -5.463E+01 -4.427E+01
 26 -1.151E+00 -2.913E+01 -4.564E+01
 27 -1.151E+00 -3.525E+01 -1.642E+01

15:31:50 04/28/86
TO CONTINUE, RETURN; TO STOP, TYPE 'S'?

JOB:CASE	DATE	LOADING CASE	TIME	PAGE
103:1 02	04/28/86	1	15:31:50	1
NP	U	V	W	THX
101	-3.407E-01	-3.247E-01	-3.047E-01	
102	-1.183E+00	-1.104E+00	-1.910E+01	
103	-2.725E+01	-2.599E+01	1.597E+02	
104	-7.271E+11	-7.680E+01	-2.163E+01	
105	-9.710E+01	-9.702E+01	-1.784E+01	
106	-5.163E+01	-5.313E+01	8.111E+02	
107	-1.708E+01	-1.705E+01	-1.155E+01	
108	-7.227E+01	-7.297E+01	-9.124E+02	
109	-3.497E+01	-3.821E+01	1.556E+01	
110	-3.042E+01	-3.299E+01	-2.321E+02	
111	-7.313E+01	-7.481E+01	-3.374E+03	
112	-1.910E+01	-1.731E+01	2.023E+01	
113	-3.810E+01	-3.696E+01	4.726E+02	
114	-4.011E+01	-4.581E+01	8.532E+02	
115	-1.189E+02	-7.143E+01	8.050E+01	
116	-5.688E+02	-9.095E+01	1.577E+01	
117	-2.102E+01	-7.143E+11	1.742E+01	
118	1.575E+01	-7.759E+01	8.779E+01	
119	-1.175E+01	-1.483E+01	8.467E+01	
120	-1.512E+02	-7.759E+01	2.682E+01	
121	3.273E+01	-3.377E+01	4.548E+01	
122	1.920E+01	-9.402E+01	8.097E+01	
123	1.724E+01	-3.677E+01	2.524E+01	
124	1.177E+01	-3.997E+01	5.299E+01	
125	4.572E+01	-1.161E+00	4.809E+01	
126	3.842E+01	-3.101E+01	4.418E+01	
127	1.876E+01	-7.620E+01	8.050E+01	
128	3.425E+01	-1.072E+00	5.122E+01	
129	3.564E+01	-3.580E+01	5.814E+01	
130	3.386E+01	-1.024E+00	8.801E+01	

TO CONTINUE, RETURN; TO STOP, TYPE 'S'?

JOB:CASE	DATE	LOADING CASE	TIME	PAGE
108:CASE2	04/28/86	1	15:31:50	1
NP	U	V	W	THX
121	8.179E+01	-1.112E+00	6.136E+01	
122	7.487E+01	-1.024E+00	6.210E+01	
123	1.009E+00	-1.087E+00	7.553E+01	
124	9.933E+01	-1.153E+00	7.151E+01	
125	7.411E+01	-1.067E+00	7.107E+01	

*? INFST

?> 1.10001\,1000

JOB:CASE	DATE	LOADING CASE	TIME	PAGE
108:CASE2	04/28/86	1	15:32:16	1
NE		STRESSES -- MIDDLE		

1	0.000E+01
2	0.000E+01
3	-1.203E+03
4	1.772E+01
5	-1.094E+04
6	1.933E+01
7	2.859E+03
8	-1.203E+03
9	1.711E+03
10	-7.133E+02
11	1.913E+03
12	2.059E+03
13	1.711E+03
14	-7.133E+02
15	1.913E+03

51 1.709E+03
52 -7.165E+02
53 4.601E+03
54 1.729E+04
55 1.709E+03
56 -7.174E+02
57 3.543E+03
58 1.701E+04
59 3.405E+03
60 1.511E+04

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOBCASE 04/28/86 15:32:34 PAGE 5
NE
51 1.709E+03
52 -7.165E+02
53 4.601E+03
54 1.729E+04
55 1.709E+03
56 -7.174E+02
57 3.543E+03
58 1.701E+04
59 3.405E+03
60 1.511E+04

04/28/86

LOADING CASE 1
STRESSES -- MIDDLE

15:32:34

PAGE 5

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOBCASE 04/28/86 15:32:34 PAGE 6
NE
51 1.709E+03
52 -7.165E+02
53 4.601E+03
54 1.729E+04
55 1.709E+03
56 -7.174E+02
57 3.543E+03
58 1.701E+04
59 3.405E+03
60 1.511E+04

04/28/86

LOADING CASE 1
STRESSES -- MIDDLE

15:32:34

PAGE 6

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:0492 04/28/86 LOADING CASE 1 15:32:49 PAGE

STRESSES -- MIDDLE

101 3.650E+08
102 -4.022E+08
103 2.999E+08
104 1.474E+08
105 3.821E+08
106 -3.715E+08
107 3.089E+08
108 -3.812E+08
109 -4.362E+08
110 4.203E+08
111 -4.203E+08
112 -4.203E+08
113 -4.203E+08
114 -4.203E+08
115 -4.203E+08
116 -4.203E+08
117 -4.110E-04
118 3.291E-03
119 6.581E-03
120 -2.962E-02

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:0492 04/28/86 LOADING CASE 1 15:32:57 PAGE 5

STRESSES -- MIDDLE

121 1.316E-02
122 -6.581E-02
123 9.872E-02
124 -5.594E-02
125 3.620E-02
126 2.369E-01
127 4.507E-02
128 1.765E-01
129 1.530E-01
130 1.514E-01
131 1.577E-01

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE2 04/28/86 LOADING CASE 1 15:38:18 PAGE 4
STRESSES -- MIDDLE

161 -1.169E+03 ←
162 -1.169E+03
163 -1.169E+03
164 -1.169E+03
165 -1.169E+03
166 -1.169E+03
167 -1.169E+03
168 -1.169E+03
169 -1.169E+03
170 -1.169E+03
171 -1.169E+03
172 -1.169E+03
173 -1.169E+03
174 -1.169E+03
175 -1.169E+03
176 -1.169E+03
177 -1.169E+03
178 -1.169E+03
179 -1.169E+03
180 -1.169E+03
181 -1.169E+03 ←
182 -4.433E+04
183 -4.433E+04
184 -4.433E+04
185 -4.433E+04
186 -4.433E+04
187 -4.433E+04
188 -4.433E+04
189 -4.433E+04
190 -4.433E+04

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CASE2 04/28/86 LOADING CASE 1 15:38:26 PAGE 5
STRESSES -- MIDDLE

181 -1.169E+03 ←
182 -4.433E+04
183 -4.433E+04
184 -4.433E+04
185 -4.433E+04
186 -4.433E+04
187 -4.433E+04
188 -4.433E+04
189 -4.433E+04
190 -4.433E+04
191 -1.169E+03

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

04/29/84 LOADING CASE A 18:38:40 8-38
STRESSES -- MIDDLE

221	-1.000E+00
222	-1.000E+00
223	-1.000E+00
224	-1.000E+00
225	-1.000E+00
226	-1.000E+00
227	-1.000E+00
228	-1.000E+00
229	-1.000E+00
230	-1.000E+00
231	-1.000E+00
232	-2.0115E+04
233	-2.1458E+03
234	-3.4288E+03
235	-6.7548E+03
236	-2.2858E+03
237	-3.8238E+03
238	-1.5248E+04
239	-1.0958E+04
240	-1.6248E+03

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:0452 04/29/86 LOADING CASE 1 15:33:53 PAGE
16 STRESSES -- MIDDLE
241 4 308E+02

171 -1.245E+01
172 -1.155E+01
173 -1.242E+01
174 -1.183E+01
175 -1.291E+01
176 -1.243E+01
177 -1.214E+01
178 -1.199E+01
179 -1.205E+01
180 -1.257E+01
181 -1.166E+01
182 -1.250E+01
183 -1.169E+01
184 -1.108E+00
185 -1.139E+02
186 -1.211E+01
187 -1.157E+01
188 -1.195E+00
189 -1.117E+00

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CH32 04/28/86 LOADING CASE 1 15:34:03 PAGE 10
STRESSES -- MIDDLE

190 -1.166E+01
191 -1.119E+01
192 -1.166E+01
193 -1.161E+01
194 -1.133E+00
195 -1.149E+00
196 -1.129E+00
197 -1.159E+01
198 -1.180E+01
199 -1.122E+00
200 -1.199E+00
201 -1.161E+00
202 -1.199E+00
203 -1.146E+01
204 -1.178E+01
205 -1.171E+00
206 -1.177E+00
207 -1.197E+00
208 -1.191E+00
209 -1.153E+00
210 -1.111E+00
211 -1.112E+00
212 -1.112E+00
213 -1.144E+00
214 -1.122E+00
215 -1.107E+01
216 -1.161E+00
217 -1.129E+00
218 -1.151E+00
219 -1.151E+00
220 -1.129E+00

TO CONTINUE, STRIKE 'RETURN'; TO STOP, TYPE 'S'?

JOB:CH32 04/28/86 LOADING CASE 1 15:34:19 PAGE 11
NE
201 -1.151E+01
202 -1.112E+01
203 -1.144E+00
204 -1.130E+00
205 -1.122E+00
206 -1.159E+00
207 -1.112E+00
208 -1.151E+00
209 -1.129E+00
210 -1.151E+00
211 -1.129E+00
212 -1.151E+00
213 -1.129E+00
214 -1.151E+00
215 -1.129E+00
216 -1.151E+00
217 -1.129E+00
218 -1.151E+00
219 -1.129E+00
220 -1.151E+00

--

1000 FT. ASL., STRIKE RETURNING TO STOP, TYPE 1912

351	1.122E+04
352	1.123E+04
353	1.124E+04
354	1.125E+04
355	1.126E+04
356	1.127E+04
357	1.128E+04
358	1.129E+04
359	1.130E+04
360	1.131E+04

24/28/84

LOADING CASE 1
STRENGTH -- MIDDLE

2024-03-03

P. 172

11. CYCLICAL, STRIKE RETURNING TO STOP, TIME = 0.0

TIME	DATE	LOADING CASE	INSTRUMENT	PAGE
0.000	04/23/86	1	15167104	1
		STRESSED -- MIDDLE		
0.000		1.000E+01		
0.000		1.000E+04		
0.000		-4.000E+03		
0.000		-2.000E+16		
0.000		1.000E+04		
0.000		2.000E+04		
+0.000				

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1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30 31 32 33 34 35 36 37 38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80 81 82 83 84 85 86 87 88 89 90 91 92 93 94 95 96 97 98 99 100 101 102 103 104 105 106 107 108 109 110 111 112 113 114 115 116 117 118 119 120 121 122 123 124 125 126 127 128 129 130 131 132 133 134 135 136 137 138 139 140 141 142 143 144 145 146 147 148 149 150 151 152 153 154 155 156 157 158 159 160 161 162 163 164 165 166 167 168 169 170 171 172 173 174 175 176 177 178 179 180 181 182 183 184 185 186 187 188 189 190 191 192 193 194 195 196 197 198 199 200 201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 216 217 218 219 220 221 222 223 224 225 226 227 228 229 230
```

ABEAM

This program, ABEAM, is the driver program to connect the nodes listed
in the file "BEAMA", below.

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Page 1

1. **What is the name of the organization?**
The National Council of Negro Women, Inc.

```

11 OPEN #OUT;"SEAMO"
12 WRITE #OUT
13 PRINT #OUT;"0,-7.5,0"
14 PRINT #OUT;"0,0,0"
15 PRINT #OUT;"0,7.5,0"
16 FOR I=10 TO 210 STEP 10
17 LET Y=-7.5+(13/400)*(X-100)
18 PRINT #OUT;X;",";Y;",";0
19 LET I=15-(13/200)*(X-100)
20 PRINT #OUT;X;",";I;",";12
21 PRINT #OUT;X;",";-1*Y;",";12
22 END

```

THEORY OF THE STATE

This is the driver program to connect the nodes listed in the file "BEAMC", below.

1940

PLATE 1940-1941-18417

100	,0,	8.15	,0
100	,0,	4.575	,0
110	,+4.55	,0	
110	,0,	5.5	
110	,0,	4.55	,0
120	,+3.925	,0	
120	,0,	7.85	
120	,0,	6.925	,0
130	,+3.8	,0	
130	,0,	7.2	
130	,0,	9.5	,0
140	,+3.275	,0	
140	,0,	6.55	
140	,0,	3.275	,0
150	,+2.35	,0	
150	,0,	5.9	
150	,0,	2.95	,0
150	,+2.325	,0	
160	,0,	6.35	
160	,0,	3.325	,0
170	,+2.3	,0	
170	,0,	4.5	
170	,0,	1.3	,0
170	,+1.375	,0	
170	,0,	3.35	
170	,0,	0.775	,0
170	,0,	0.75	,0

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卷之三十一

（三）在“五·四”運動中，學生們的抗議行動，是對當時社會不平等現象的一次強烈的抗擊。

卷之三十一

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第二章 第二節 一、民族主義：自覺與不自覺

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卷之三十一

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1997-1998 学年第二学期期中考试卷

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卷之三十一

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卷之三

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For more information about the study, contact Dr. Michael J. Koenig at (314) 362-3222 or via e-mail at koenig@artsci.wustl.edu.

```

1000 PRINT #OUT1;"BEAMC"
1010 PRINT #OUT1;"FROM, BRC"
1020 PRINT #OUT1;"BUTT"
1030 PRINT #OUT1;"BEAM(A)"
1040 PRINT #OUT1;"PROC, FF"
1050 PRINT #OUT1;"#CUTS"
1060 PRINT #OUT1;"STRUCTURE"
1070 PRINT #OUT1;"1000,2912"
1080 PRINT #OUT1;"A BULKM DRIVER FILE"
1090 PRINT #OUT1;"#POINT"
1100 FOR A=1 TO 59
1110 INPUT #OUT2;BEAM(A,1),BEAM(A,2),BEAM(A,3)
1120 NEXT A
1130 FOR A=1 TO 59
1140 LET TEMP1=A
1150 PRINT #OUT1;A
1160 LET TEMP(A,2)=BEAM(A,2)
1170 LET TEMP(A,1)=BEAM(A,3)+44.5099
1180 LET TEMP(A,3)=BEAM(A,1)*(-1)-10
1190 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1200 PRINT #OUT1;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1210 NEXT A
1220 FOR A=1 TO 59
1230 LET TEMP(A,3)=-1+TEMP(A,3)
1240 LET TEMP(A,2)=TEMP(A,3)
1250 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1260 NEXT A
1270 FOR A=1 TO 59
1280 LET TEMP(A,4)=TEMP(A,2)
1290 LET TEMP(A,2)=TEMP(A,3)
1300 LET TEMP(A,3)=TEMP(A,4)
1310 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1320 NEXT A
1330 FOR A=1 TO 59
1340 LET TEMP(A,2)=-1*TEMP(A,2)
1350 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1360 NEXT A
1370 LET ANG=COS(PI/4)
1380 FOR A=1 TO 59
1390 LET TEMP(A,4)=TEMP(A,2)
1400 LET TEMP(A,2)=TEMP(A,2)*ANG+TEMP(A,3)*ANG
1410 LET TEMP(A,3)=TEMP(A,5)*ANG+TEMP(A,3)*ANG
1420 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1430 NEXT A
1440 FOR A=1 TO 59
1450 LET TEMP(A,3)=-1*TEMP(A,3)
1460 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1470 NEXT A
1480 FOR A=1 TO 59
1490 LET TEMP(A,2)=-1*TEMP(A,2)
1500 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1510 NEXT A
1520 FOR A=1 TO 59
1530 LET TEMP(A,3)=-1*TEMP(A,3)
1540 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1550 NEXT A
1560 FOR A=1 TO 59
1570 LET TEMP(A,2)=TEMP(A,1)
1580 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1590 NEXT A
1600 FOR A=1 TO 59
1610 LET TEMP(A,3)=TEMP(A,1)
1620 PRINT #OUT3;TEMP(A,1);";";TEMP(A,2);";";TEMP(A,3)
1630 NEXT A
1640 PRINT #OUT1;"BEAMC"
1650 FOR A=1 TO 59

```



```

1500 IF A=0 THEN PRINT #OUT1;"L";STR$(59);STR$(57)
1510 IF A=0 THEN PRINT #OUT1;"L";STR$(59);STR$(57)
1520 PRINT #OUT3;"2,";59+A*59;"";57+A*59
1530 PRINT #OUT3;"2,";59+A*59;"";57+A*59
1540 PRINT #OUT1;"L";STR$(B);STR$(B+3)
1550 IF A=0 THEN PRINT #OUT1;"B;"";B+3";2"
1560 PRINT #OUT3;"2,";B+3+A*59;"";B+3+A*59
1570 IF A=0 THEN PRINT #OUT1;"L";STR$(B+1);STR$(B+4)
1580 IF A=0 THEN PRINT #OUT1;"B+1";";B+4";2"
1590 PRINT #OUT3;"2,";B+1+A*59;"";B+4+A*59
1600 IF A=0 THEN PRINT #OUT1;"L";STR$(B+2);STR$(B+4)
1610 IF A=0 THEN PRINT #OUT1;"B+2";";B+4";2"
1620 PRINT #OUT3;"2,";B+2+A*59;"";B+4+A*59
1630 IF A=0 THEN PRINT #OUT1;"L";STR$(B+2);STR$(B+3)
1640 IF A=0 THEN PRINT #OUT1;"B+2";";B+3";2"
1650 PRINT #OUT3;"2,";B+2+A*59;"";B+3+A*59
1660 NEXT B
1670 FOR B=3 TO 54 STEP 3
1680 IF A=0 THEN PRINT #OUT1;"L";STR$(B);STR$(B+3)
1690 IF A=0 THEN PRINT #OUT1;"B;"";B+3";2"
1700 PRINT #OUT3;"2,";B+3+A*59;"";B+3+A*59
1710 NEXT B
1720 FOR B=1 TO 55 STEP 6
1730 IF A=0 THEN PRINT #OUT1;"L";STR$(B);B+4
1740 IF A=0 THEN PRINT #OUT1;"B;"";B+4";2"
1750 PRINT #OUT3;"2,";B+A*59;"";B+4+A*59
1760 NEXT B
1770 FOR B=5 TO 53 STEP 6
1780 IF A=0 THEN PRINT #OUT1;"L";STR$(B);STR$(B+7)
1790 IF A=0 THEN PRINT #OUT1;"B;"";B+2";2"
1800 PRINT #OUT3;"2,";B+A*59;"";B+2+A*59
1810 NEXT B
1820 FOR B=4 TO 55 STEP 9
1830 IF A=0 THEN PRINT #OUT1;"L";STR$(B);STR$(B+1)
1840 IF A=0 THEN PRINT #OUT1;"B;"";B+1";2"
1850 PRINT #OUT3;"2,";B+A*59;"";B+1+A*59
1860 NEXT B
1870 FOR B=5 TO 54 STEP 3
1880 IF A=0 THEN PRINT #OUT1;"L";STR$(B);STR$(B+1)
1890 IF A=0 THEN PRINT #OUT1;"B;"";B+1";2"
1900 PRINT #OUT3;"2,";B+A*59;"";B+A*59+1
1910 NEXT B
1920 NEXT A
1930 FOR A=0 TO 7
1940 IF A=0 THEN PRINT #OUT1;"L6061"
1950 IF A=0 THEN PRINT #OUT1;"60,61,2"
1960 PRINT #OUT3;"2,";473+A*66;"";474+A*66
1970 IF A=0 THEN PRINT #OUT1;"L6162"
1980 IF A=0 THEN PRINT #OUT1;"61,62,2"
1990 PRINT #OUT3;"2,";474+A*66;"";475+A*66
2000 IF A=0 THEN PRINT #OUT1;"L6064"
2010 IF A=0 THEN PRINT #OUT1;"60,64,2"
2020 PRINT #OUT3;"2,";473+A*66;"";477+A*66
2030 IF A=0 THEN PRINT #OUT1;"L6164"
2040 IF A=0 THEN PRINT #OUT1;"61,64,2"
2050 PRINT #OUT3;"2,";474+A*66;"";477+A*66
2060 IF A=0 THEN PRINT #OUT1;"L6264"
2070 IF A=0 THEN PRINT #OUT1;"62,64,2"
2080 PRINT #OUT3;"2,";475+A*66;"";477+A*66
2090 FOR B=60 TO 120 STEP 3

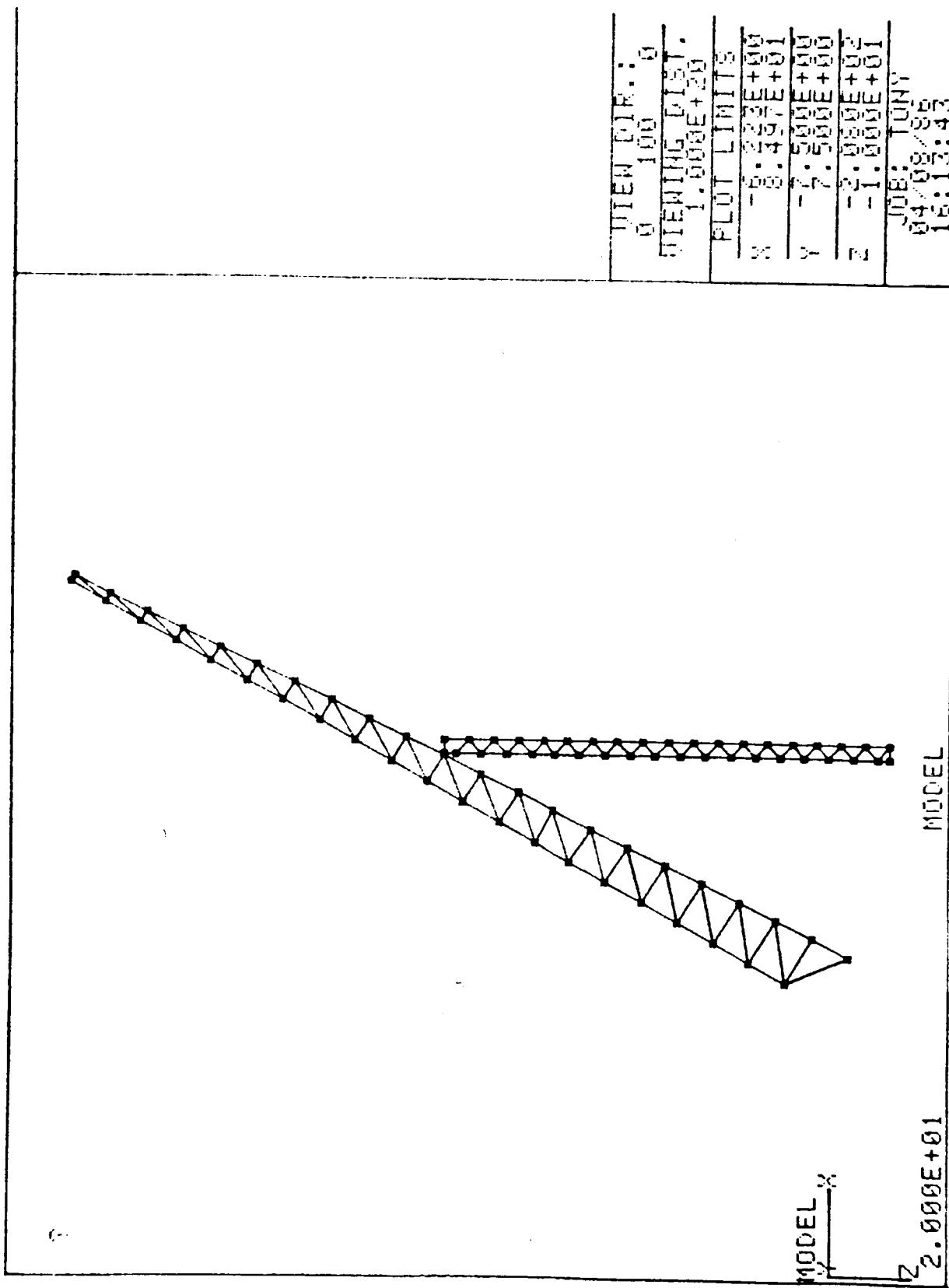
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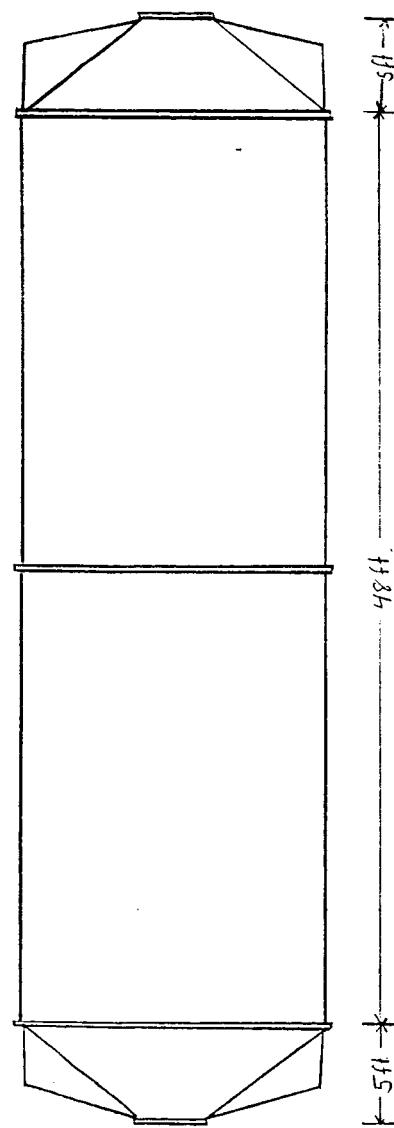
1000 PRINT #OUT1;"2,";B+413+A*B66;"";STR$(B+2);STR$(B+3)
1010 PRINT #OUT1;"L";STR$(B+2);STR$(B+3)
1020 PRINT #OUT1;"B";"1";B+5;"2"
1030 PRINT #OUT1;"2,";B+413+A*B66;"";B+413+A*B66
1040 PRINT #OUT1;"L";STR$(B+2);STR$(B+3)
1050 PRINT #OUT1;"B";"1";B+5;"2"
1060 PRINT #OUT1;"2,";B+413+A*B66;"";B+413+A*B66
1070 PRINT #OUT1;"L";STR$(B+2);STR$(B+3)
1080 PRINT #OUT1;"B";"1";B+5;"2"
1090 PRINT #OUT1;"B";"1";B+5;"2"
1100 PRINT #OUT1;"B";"1";B+5;"2"
1110 PRINT #OUT1;"B";"1";B+5;"2"
1120 PRINT #OUT1;"B";"1";B+5;"2"
1130 PRINT #OUT1;"B";"1";B+5;"2"
1140 PRINT #OUT1;"B";"1";B+5;"2"
1150 PRINT #OUT1;"B";"1";B+5;"2"
1160 PRINT #OUT1;"B";"1";B+5;"2"
1170 PRINT #OUT1;"B";"1";B+5;"2"
1180 PRINT #OUT1;"B";"1";B+5;"2"
1190 PRINT #OUT1;"B";"1";B+5;"2"
1200 PRINT #OUT1;"B";"1";B+5;"2"
1210 PRINT #OUT1;"B";"1";B+5;"2"
1220 PRINT #OUT1;"B";"1";B+5;"2"
1230 PRINT #OUT1;"B";"1";B+5;"2"
1240 PRINT #OUT1;"B";"1";B+5;"2"
1250 PRINT #OUT1;"B";"1";B+5;"2"
1260 PRINT #OUT1;"B";"1";B+5;"2"
1270 PRINT #OUT1;"B";"1";B+5;"2"
1280 PRINT #OUT1;"B";"1";B+5;"2"
1290 PRINT #OUT1;"B";"1";B+5;"2"
1300 PRINT #OUT1;"B";"1";B+5;"2"
1310 PRINT #OUT1;"B";"1";B+5;"2"
1320 PRINT #OUT1;"B";"1";B+5;"2"
1330 PRINT #OUT1;"B";"1";B+5;"2"
1340 PRINT #OUT1;"B";"1";B+5;"2"
1350 PRINT #OUT1;"B";"1";B+5;"2"
1360 PRINT #OUT1;"B";"1";B+5;"2"
1370 PRINT #OUT1;"B";"1";B+5;"2"
1380 PRINT #OUT1;"B";"1";B+5;"2"
1390 PRINT #OUT1;"B";"1";B+5;"2"
1400 PRINT #OUT1;"B";"1";B+5;"2"
1410 PRINT #OUT1;"B";"1";B+5;"2"
1420 PRINT #OUT1;"B";"1";B+5;"2"
1430 PRINT #OUT1;"B";"1";B+5;"2"
1440 PRINT #OUT1;"B";"1";B+5;"2"
1450 PRINT #OUT1;"B";"1";B+5;"2"
1460 PRINT #OUT1;"/"
1470 PRINT #OUT1;"ELMAT,2"
1480 PRINT #OUT1;"1"
1490 PRINT #OUT1;"6.0E4,1.0E7"
1500 PRINT #OUT1;"/"
1510 PRINT #OUT1;"ETH,1"
1520 REM*****REMOVED*****
1530 PRINT #OUT1;".5"
1540 PRINT #OUT1;"/"
1550 PRINT #OUT1;"QUIT"
1560 END

```

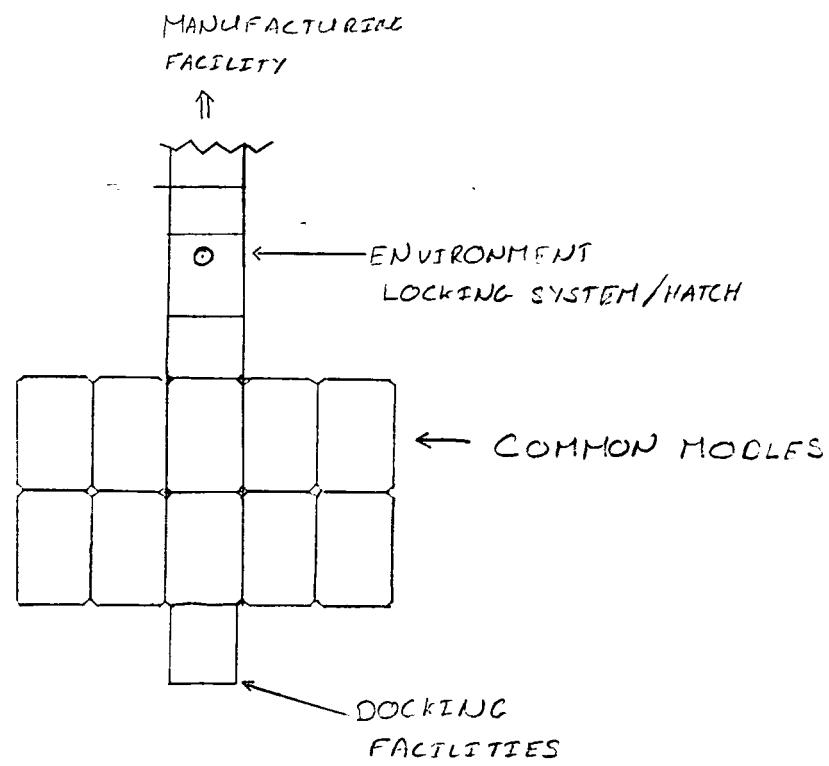
Ready

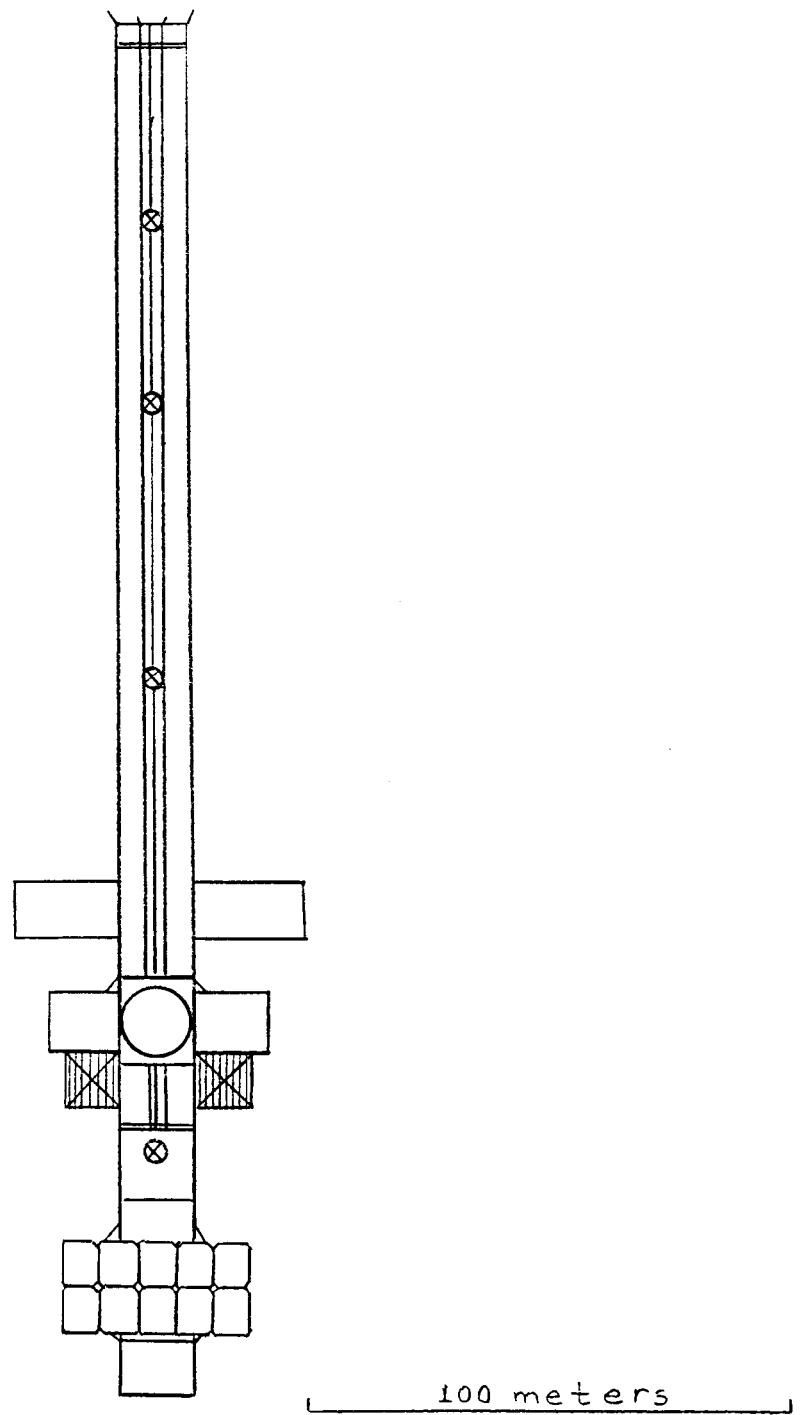


CHIMON HOLLOW

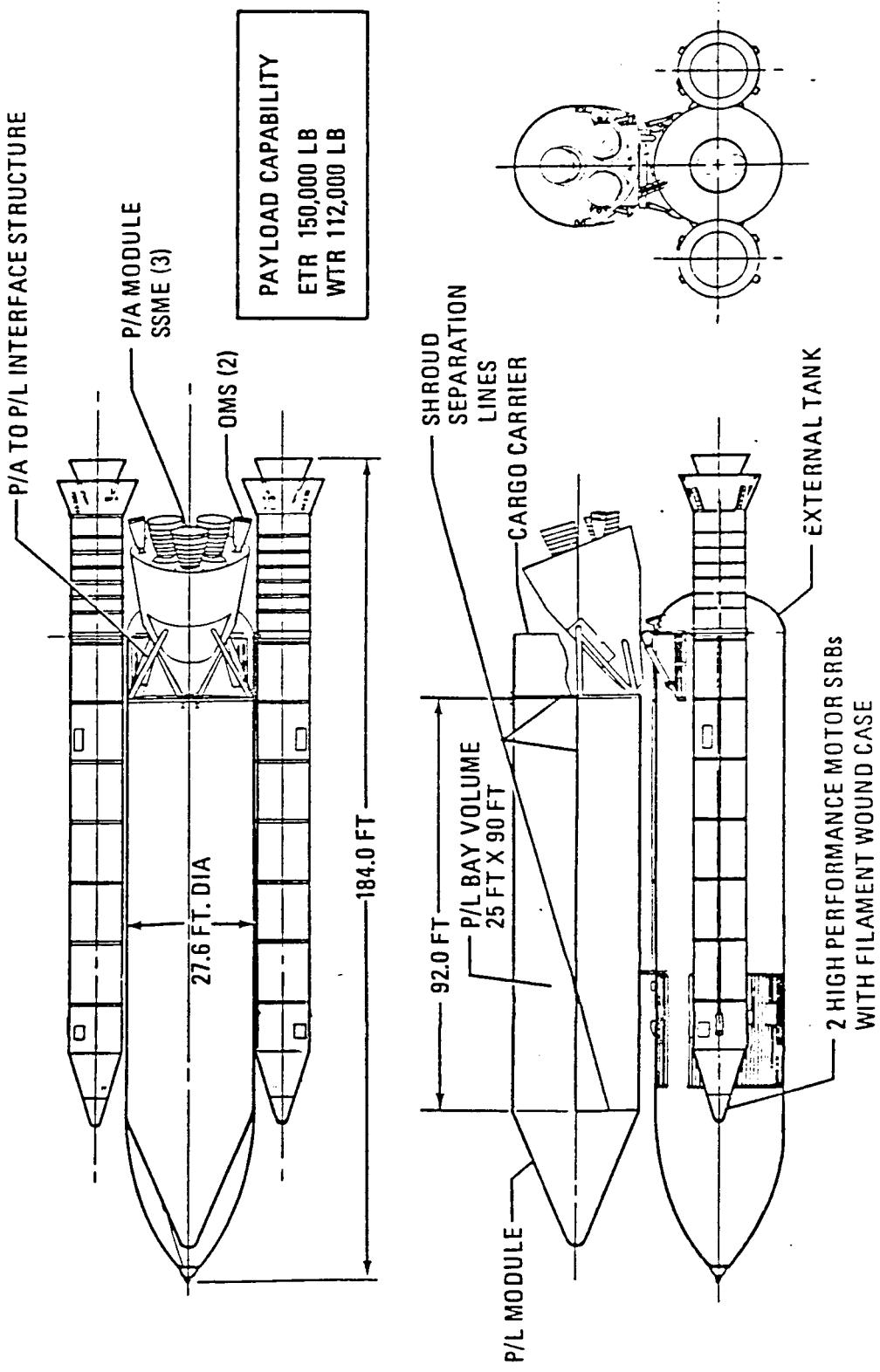


115 ft - 484 ft - 54 ft

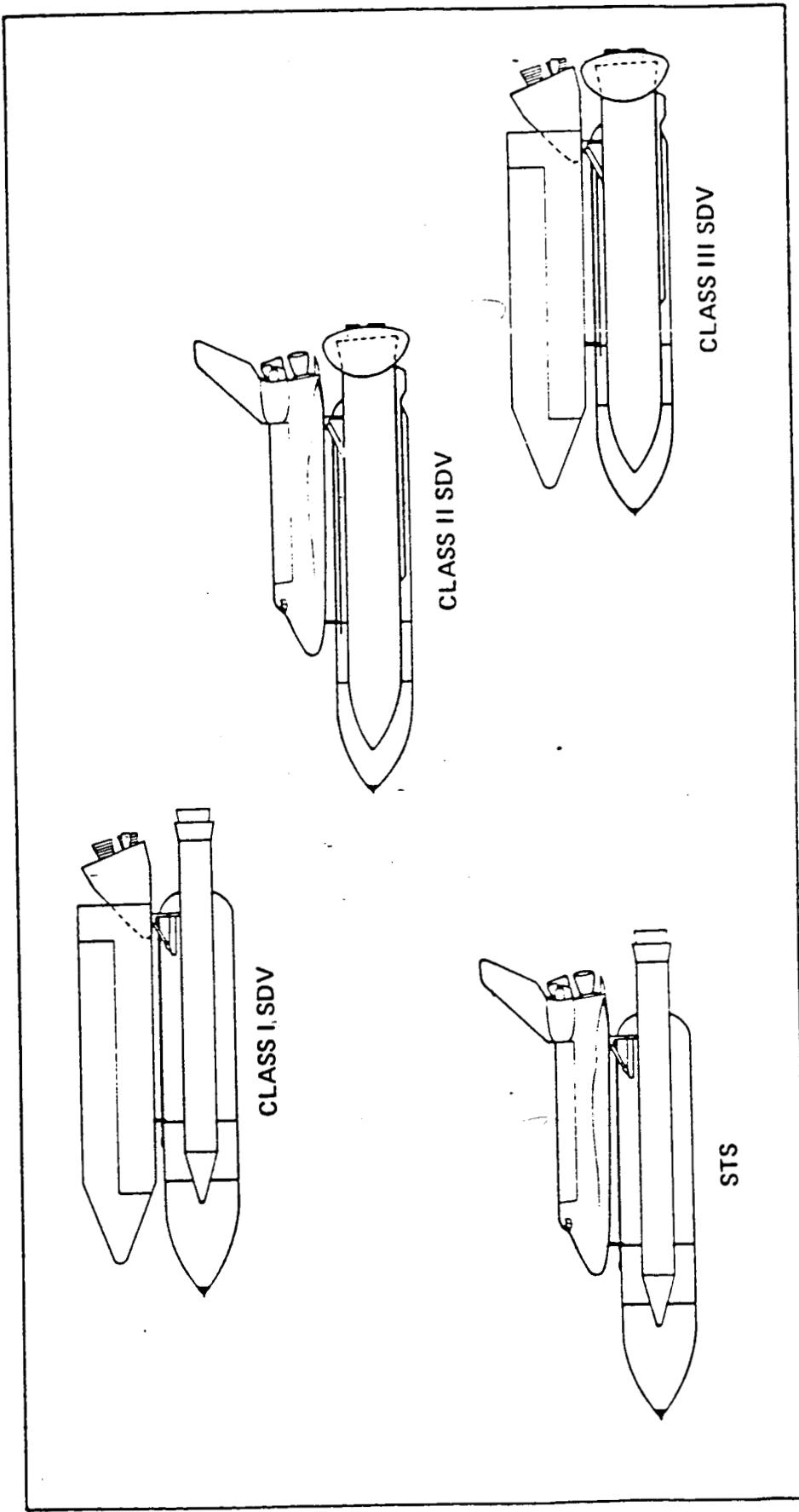




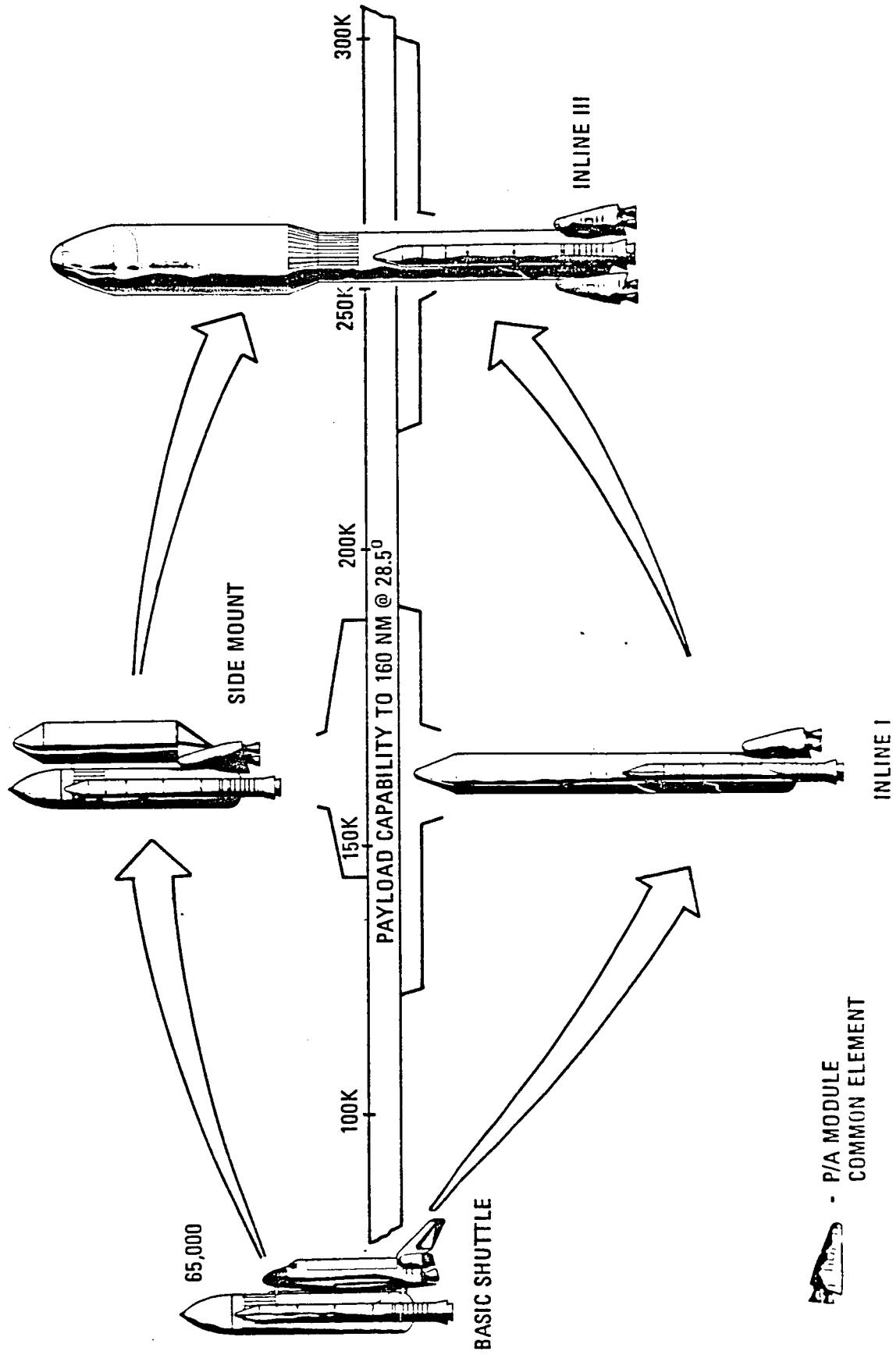
RLV Sidemount Configuration

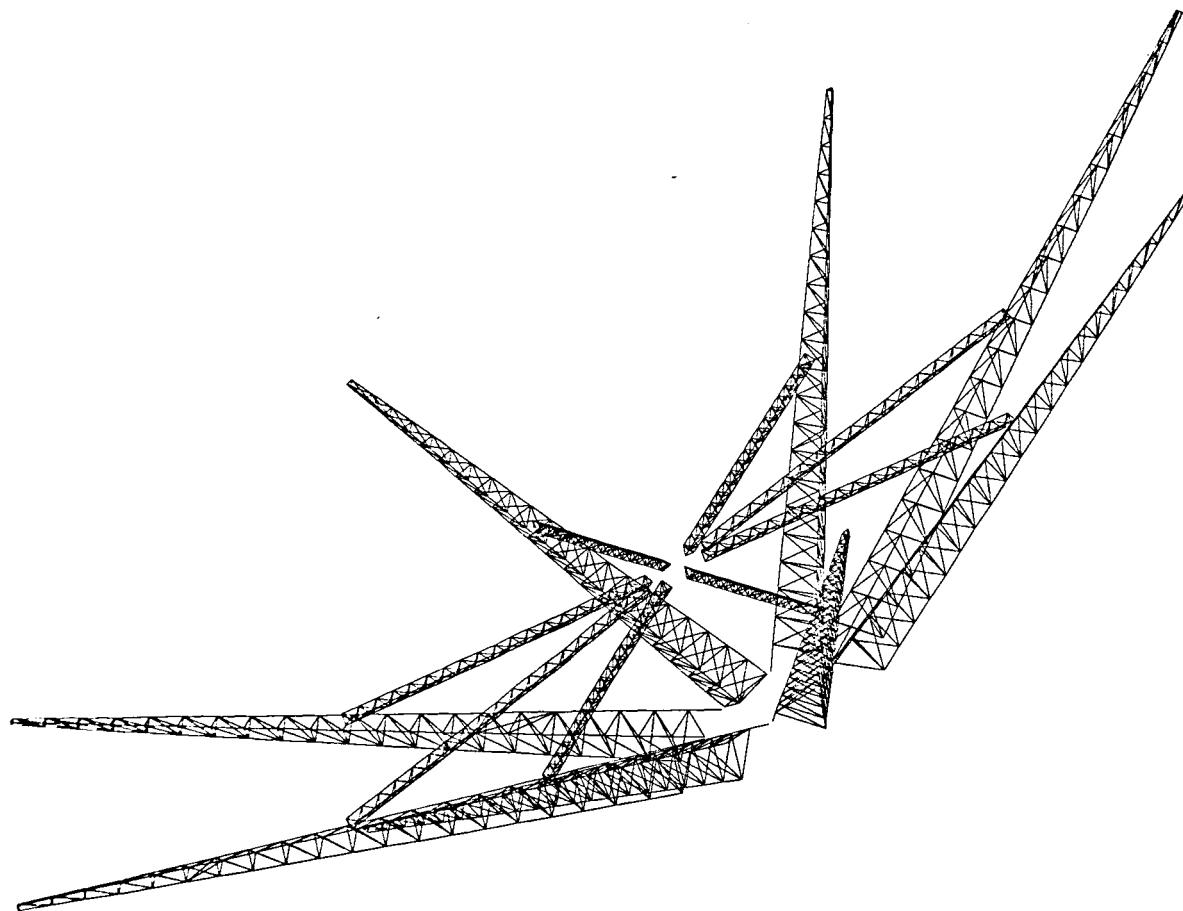


STS AND PHASE II BASELINE VEHICLE CLASSES



Development Options





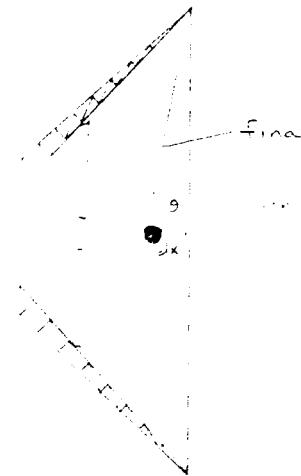
APPENDIX E

Force Calculations

The means with which the mass is to be stopped is similar to an aircraft carrier. The energy relation:

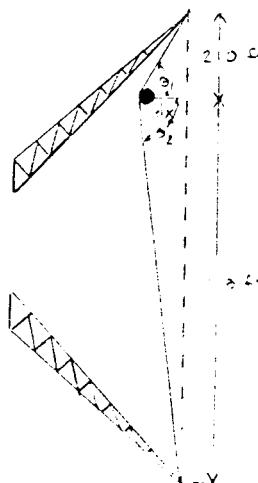
$$K_i + U = K_f$$

applies, where $K_f = 0$



$$U = -2 \int_0^x T \cos \theta \, dx$$

Since the allowable distance which the mass may travel on deceleration is restricted more for off-center catches, this is the case to be examined.



dx is limited to 100 ft
to completely model the entire
system, this model must be X4

$$U = 4 \int_0^x T \cos \theta_1 \, dx + 4 \int_0^x T \cos \theta_2 \, dx$$

$$\cos \theta_1 = \frac{x}{\sqrt{x^2 + (210^2)}} \quad \cos \theta_2 = \frac{x}{\sqrt{x^2 + (1113)^2}}$$

$$dx = 100 \text{ ft}$$

$$\text{target area dia.} = 908 \text{ ft}$$

$$T = \text{constant}$$

$$\begin{aligned} U &= -4 \int_0^{100} T \frac{x}{\sqrt{x^2 + z^2}} dx + 4 \int_0^{100} T \frac{x}{\sqrt{x^2 + 1116^2}} dx \\ &= -4T \left[\sqrt{x^2 + z^2} \right]_0^{100} + -4T \left[\sqrt{x^2 + 1116^2} \right]_0^{100} \\ &= -4T \left[\sqrt{100^2 + z^2} - z \right] + -4T \left[\sqrt{100^2 + 1116^2} - 1116 \right] \\ &= -4T \left[(22.5041) + -(4.4638) \right] \\ &= -T (108.2) \end{aligned}$$

$$K_i = U = 108.2 T \quad V = 200 \text{ m/s} = 642 \text{ ft/s}$$

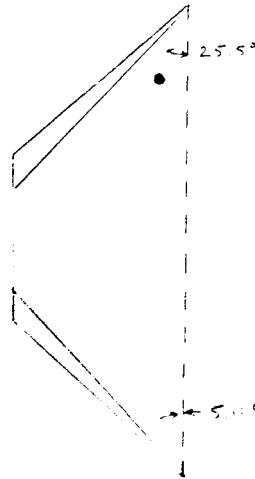
$$1/2 m V^2 = 108.2 T$$

$$m = \{2(108.2)T\}/(642)^2$$

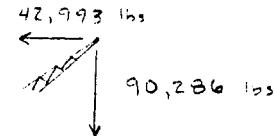
$$m = (5.2558 \times 10^{-4})$$

$$\begin{aligned} T &= 100,000 \text{ lbs.} \quad m = 52.558 \text{ slugs} \\ &\quad = 768.5 \text{ kg} \end{aligned}$$

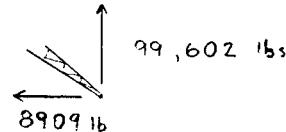
The loads will be applied in the following manner:



CASE 1- off center hit, upper



CASE 2- off center hit, lower



Notes

¹G.W. Driggers and J.E. Newman, "Establishment of a Space Manufacturing Facility," Space Based Manufacturing From Nonterrestrial Materials, New York, p. 148

²Driggers and Newman, p.146.

³Ivan Bekey, Space Stations and Space Platforms-Concepts, Design, Infrastructure, and Uses, New York, p.298.

⁴Driggers and Newman, p.150.

⁵Ivan Bekey, "Permanent Presence-Making it Work", 22nd Goddard Memorial Symposium, San Diego, p. 138.

⁶Lecture notes from Professor McCandless'es class.

References

- Bekey, Ivan. "Permanent Presence-Making it Work." 22nd Goddard Memorial Symposium, March 1984, pp.73-111.
- Bekey, Ivan, ed. Space Stations and Space Platforms-Concepts, Design Infrastructure and Uses. New York: American Institute of Aeronautics and Astronautics, 1985.
- O'Neill, Gerald K., and B. O'Leary, eds. Space-Based Manufacturing From Nonterrestrial Materials. New York: American Institute of Aeronautics and Astronautics, 1977.
- Space World, "Big Dumb Booster" March 1983, p.33.

* We also used class notes from Professor McCandless's lectures.